



# Initiation of Hydrogen Detonation

Andrzej Teodorczyk  
*Warsaw University of Technology*



# Onset of Detonation



## Direct initiation by rapid deposition of energy

- Blast wave
- Shock reflection
- Shock focusing

## Indirect initiation through the acceleration of a series of progressively more reactive states

- SWACER - Synchronized initiation
  - Flash Photolysis
  - Turbulent mixing
- DDT – Deflagration to Detonation Transition



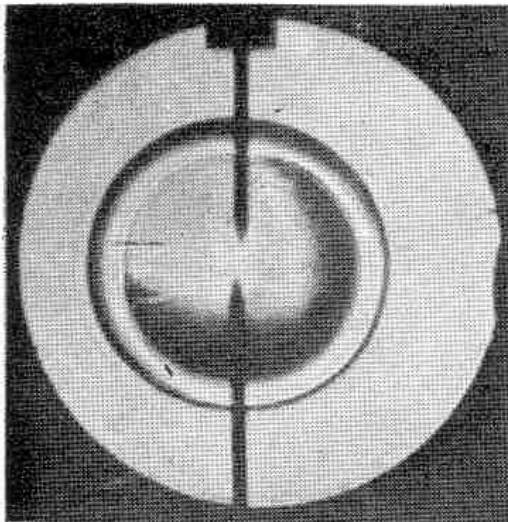
# Outline



- Direct initiation by rapid energy release
- Direct initiation by shock reflection
- Direct initiation by shock focusing
- SWACER – synchronized initiation
- DDT – Deflagration to Detonation Transition

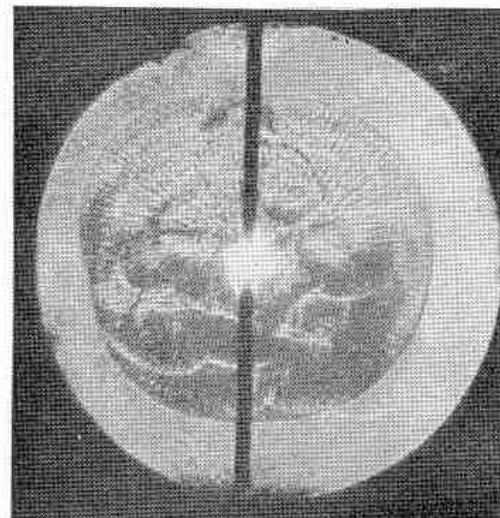
Stoichiometric hydrogen-chlorine mixture; a) subcritical; b) critical; c) supercritical

a)



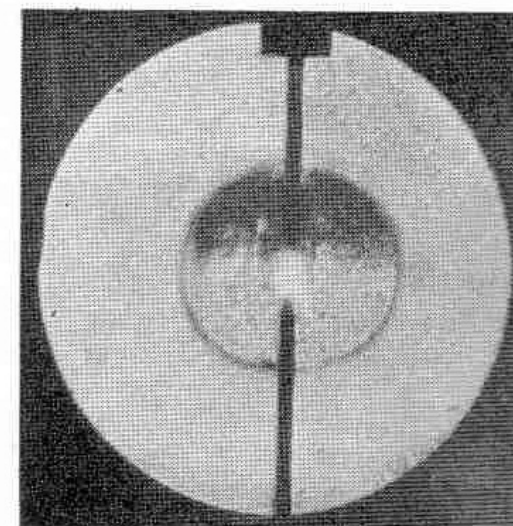
$p_0 = 0.4 \text{ bar}$   
 $E_0 = 3.6 \text{ J}$

b)



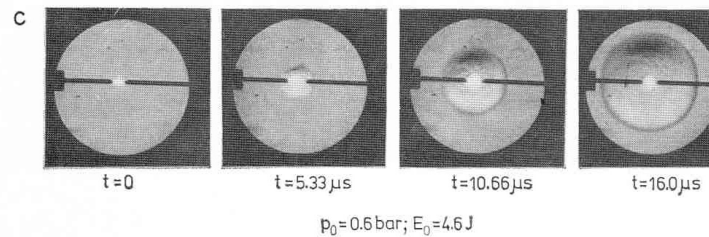
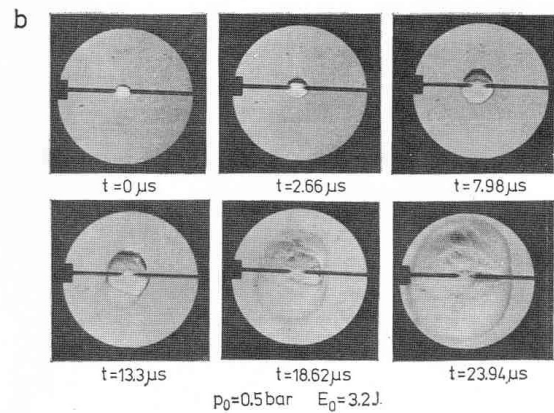
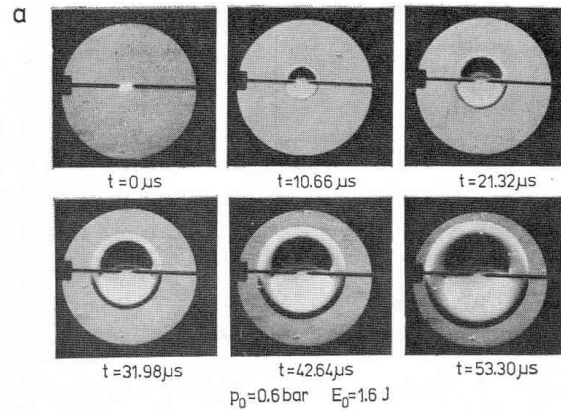
$p_0 = 0.4 \text{ bar}$   
 $E_0 = 4.1 \text{ J}$

c)



$p_0 = 0.4 \text{ bar}$   
 $E_0 = 4.7 \text{ J}$

*(Teodorczyk et al. 1978)*



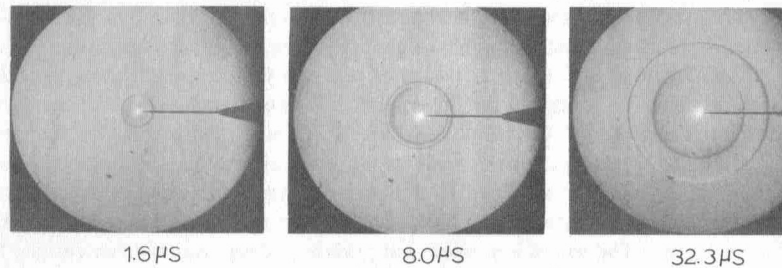
Stoichiometric hydrogen-chlorine mixture;

a) subcritical;

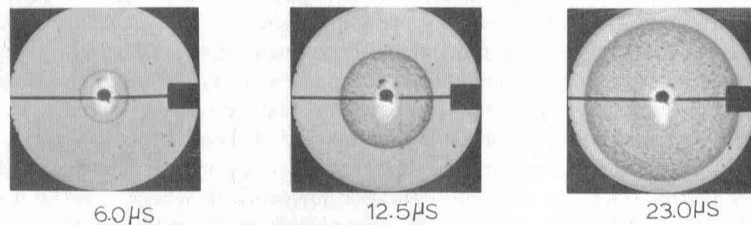
b) critical;

c) supercritical

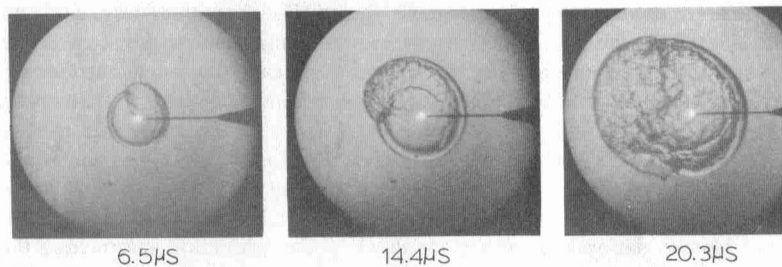
(Source: Teodorczyk et al. 1978)



a. SUBCRITICAL : 80 Torr  $2C_2H_2+5O_2$ ; Igniter: LASER SPARK



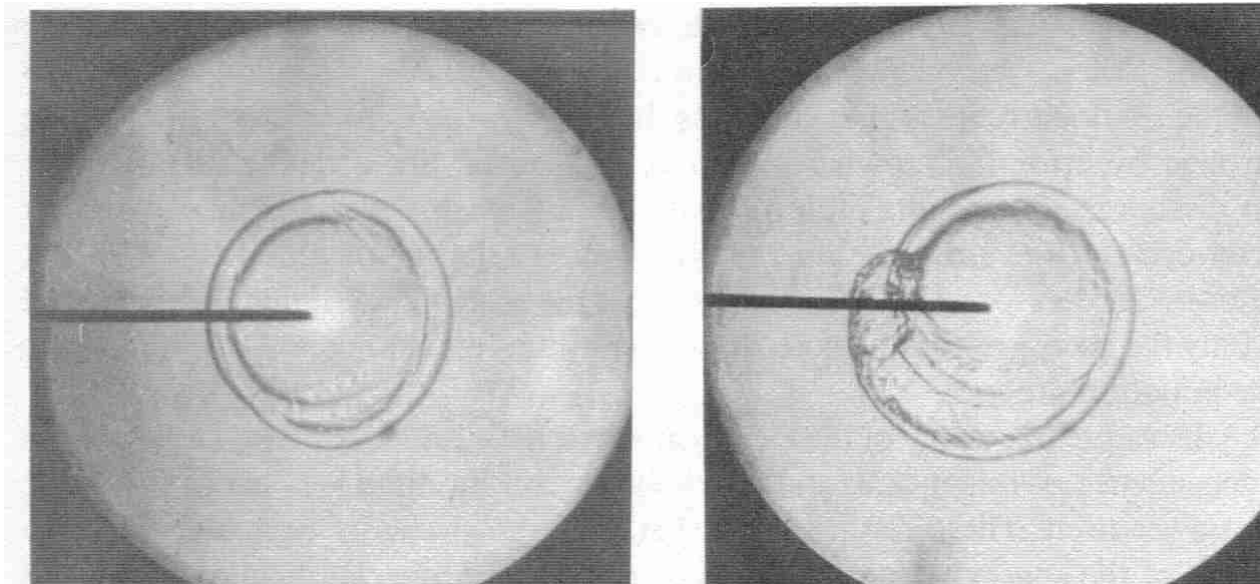
b. SUPERCRITICAL: 120 Torr  $H_2+Cl_2$ ; Igniter: ELECTRICAL SPARK



c. CRITICAL: 100 Torr  $2C_2H_2+5O_2$ ; Igniter: LASER SPARK

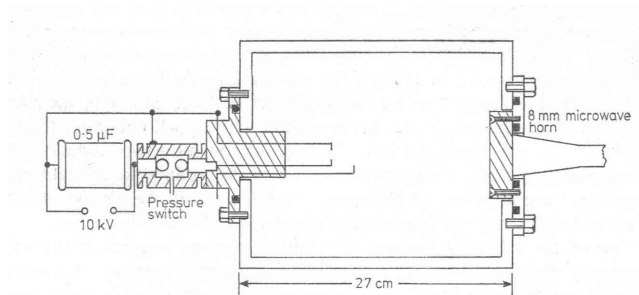
- a) subcritical; acetylene-oxygen; laser spark
- b) supercritical; hydrogen-chlorine, electric spark
- c) critical; acetylene-oxygen, laser spark

(Source: Lee et al. 1977)



Onset of detonation at the end of the quasi-steady period of the critical regime. Stoichiometric  $C_2H_2-O_2$  at 100 torr, ignition by laser spark

(Source: Lee et al. 1977)

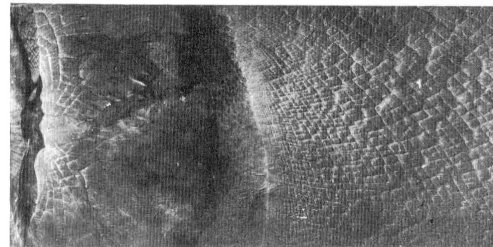


*Smoked foil records of initiation of spherical detonations.  
Exploding wire, 25 J energy  
 $2C_2H_2+5O_2+3Ar$*

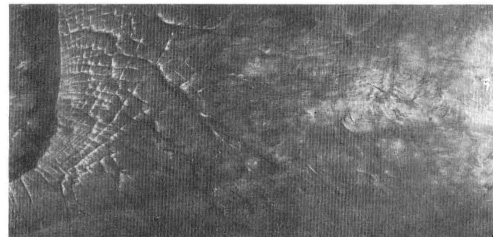
- (a) 60 Torr;*
- (b) 55 Torr;*
- (c) 51 Torr, wave failure occurs.*
- (d) lead azide source, 31 Torr;*
- (e) details of critical initiation using exploding wire*



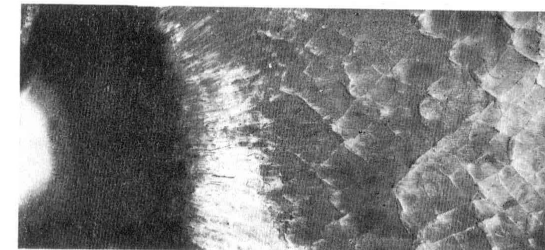
(a)



(b)



(c)



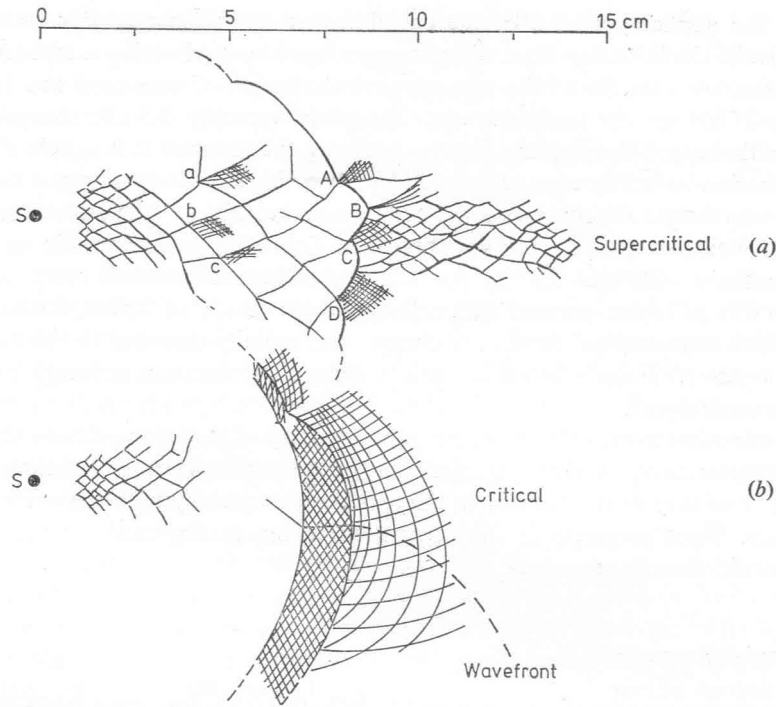
(d)



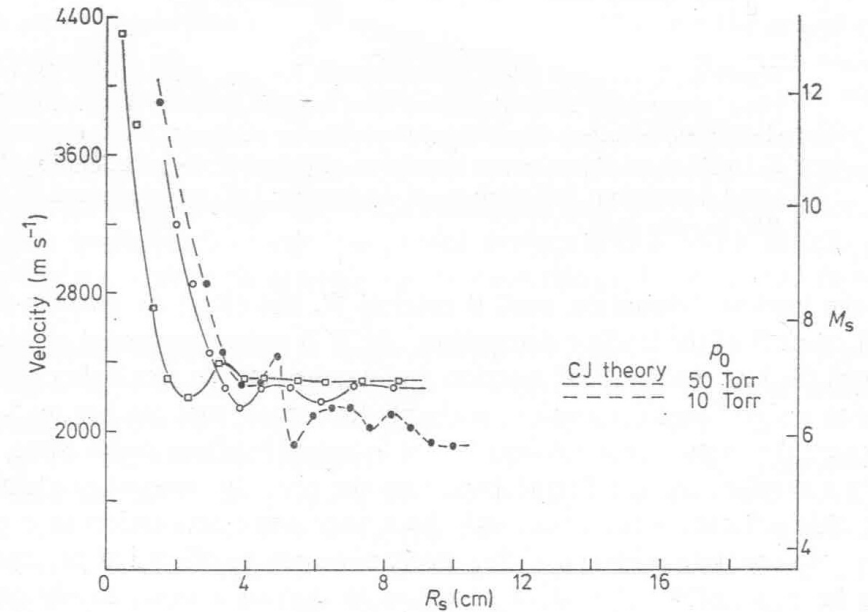
(e)

(Source: Edwards et al. 1978)



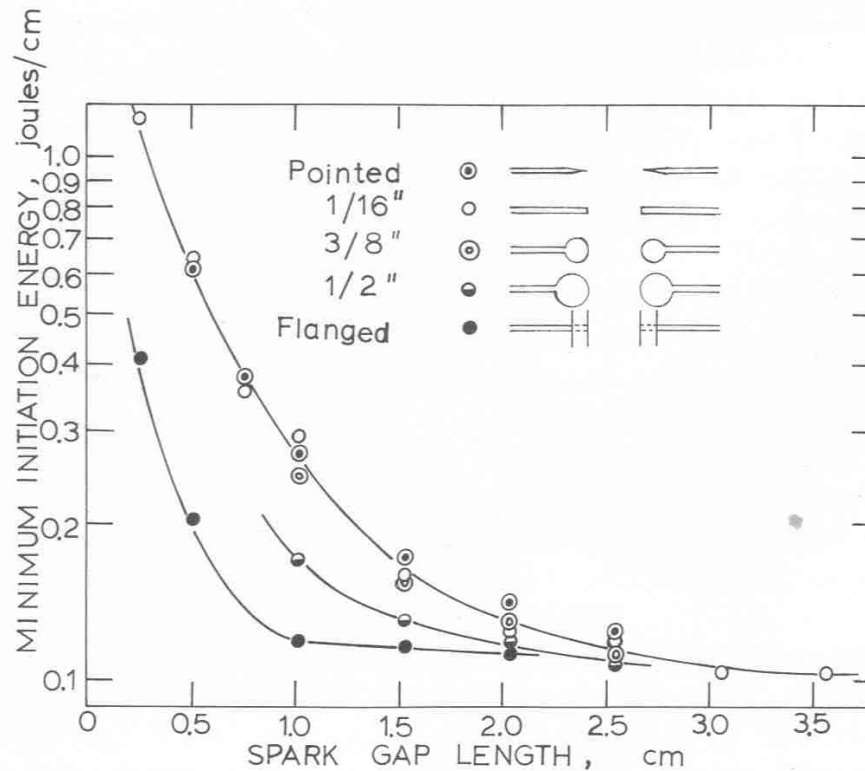


(a) supercritical and (b) critical initiation, using exploding wire;



45 mg lead azide. Initial pressure,  $p_0$ : ( ) 50 Torr; (o) 30 Torr; (•) 15 Torr,

(Source: Edwards et al. 1978)

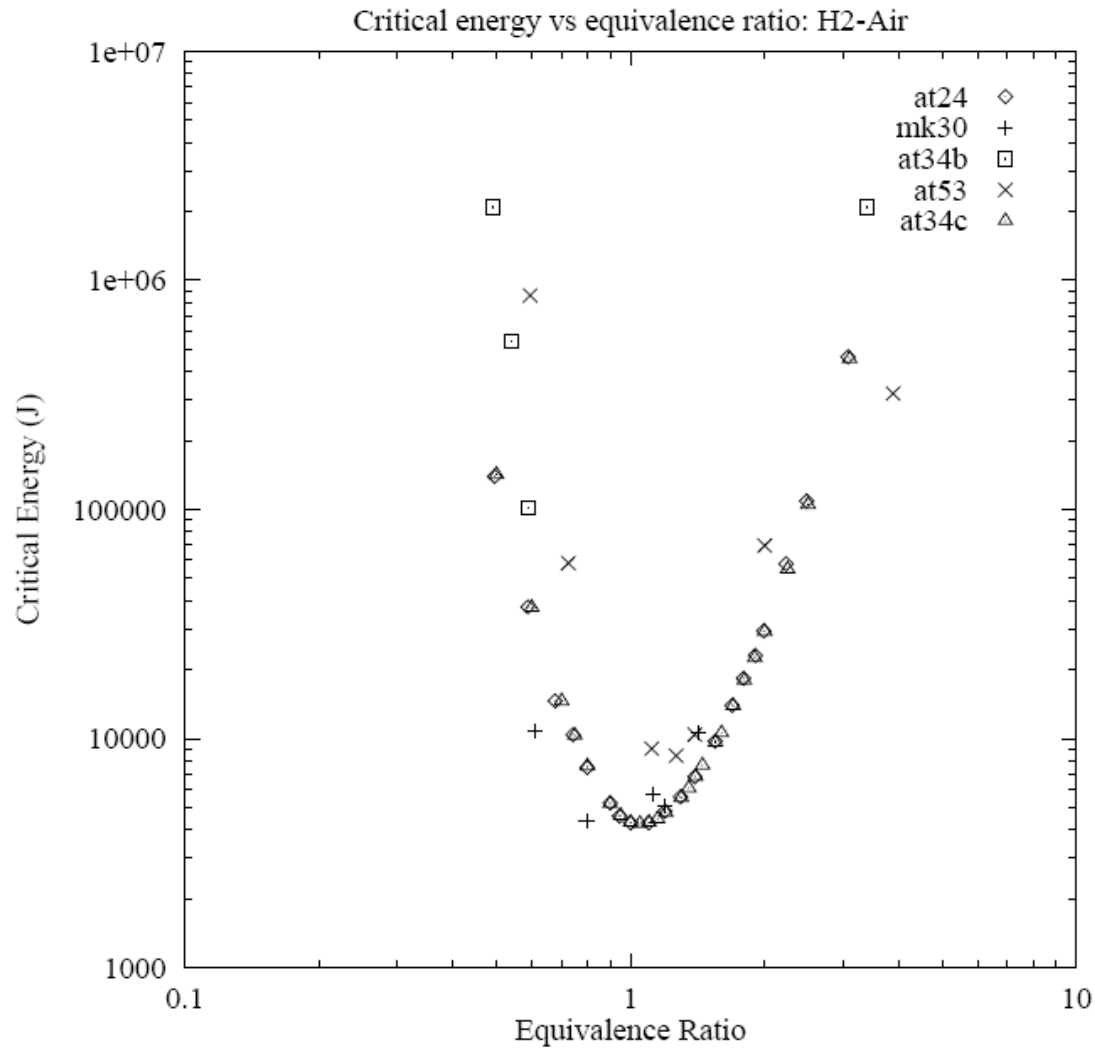


*critical energy for direct initiation of spherical detonations in stoichiometric C<sub>2</sub>H<sub>2</sub>-O<sub>2</sub> at 100 torr*

(Source: Matsui & Lee, 1976)



# Direct initiation of detonation



From detonation database



# Detonation database



**Detonation Database**

[Accessing the Data](#) | [Summary Graphs](#) | [Data Sets](#) | [References](#) | [Database Search](#)

### Abstract

Welcome to the GALCIT Explosion Dynamics Laboratory Detonation Database. The goal of this project is to compile, catalog and present experimental data on gaseous detonations. These data currently include cell width, critical tube diameter, initiation energy, and minimum tube diameter. They are formatted in tables and summary graphs, with citations to the original references. A printed version and a World Wide Web version have been prepared. The purpose of this database is to facilitate explosion hazards evaluations and comparisons with numerical simulations of detonation behavior.

**Introduction** to the Detonation Database project.

**Contributors.** Authors of the database.

**Disclaimer.** We're not perfect.

**Citations.** Using the data in publications.

**How to Access the Data.** Some useful information and tips.

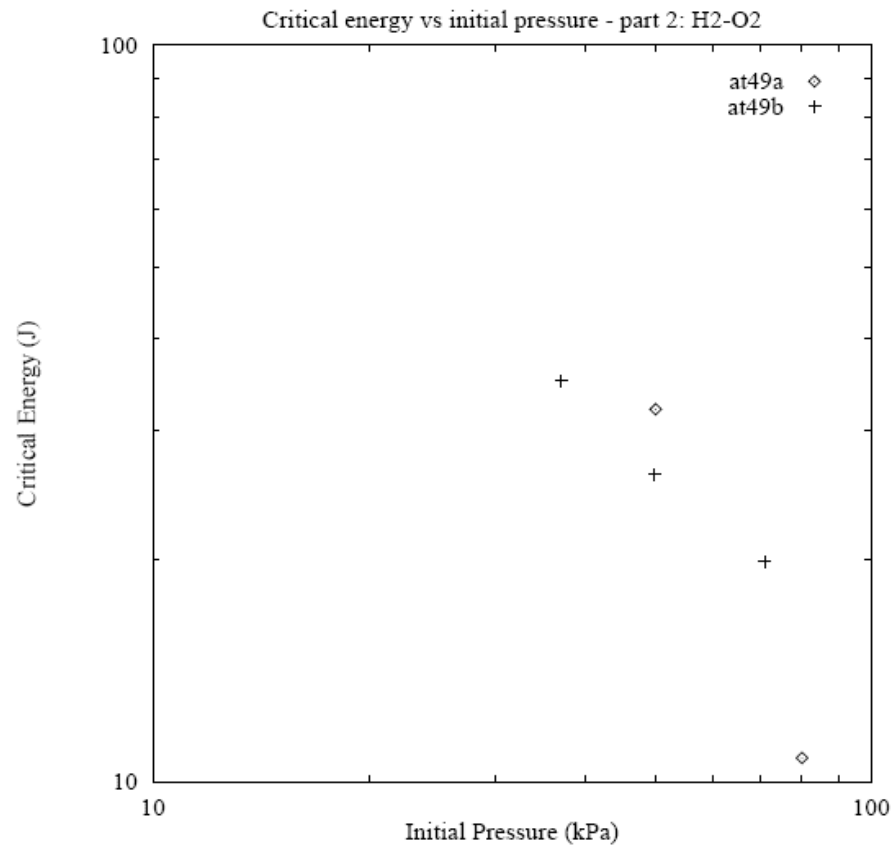
**How the Database Works.** For those who are interested.

**The Database.** Links to the different branches.

*Edited Last: Jan 29, 2005*  
*Joe Shepherd*

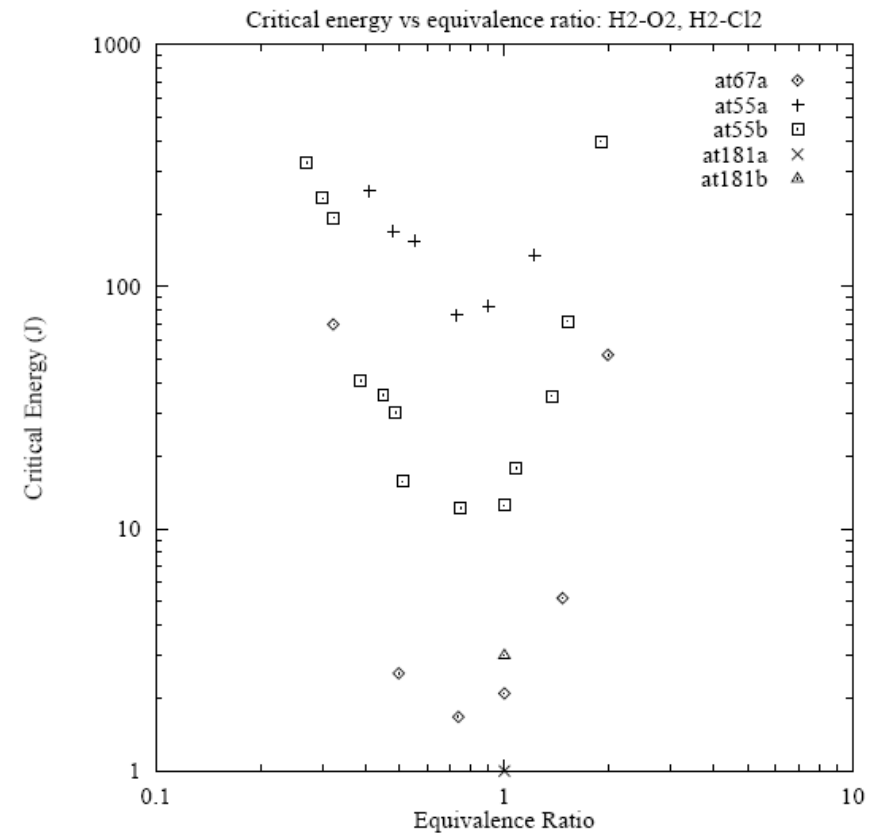


# Direct initiation of detonation



at49a: T=293 K;

at49b: T=123 K;





# Direct initiation of detonation



Critical energy of initiation:

$$E_c = \frac{2197}{16} \pi \rho_0 V_{CJ}^2 I \lambda^3$$

$\rho_0$  - initial density of the mixture

$V_{CJ}$  - Chapman-Jouget detonation velocity

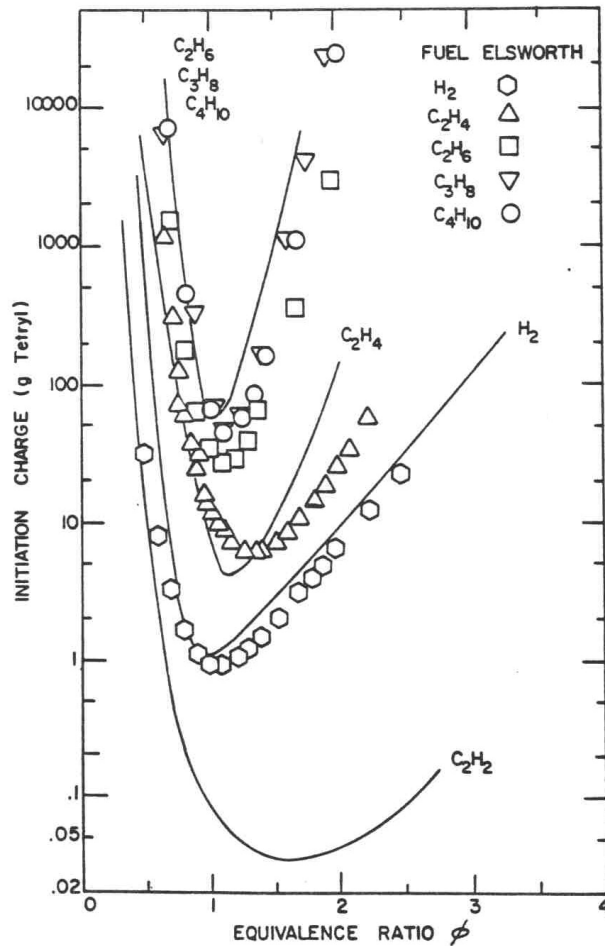
$\lambda$  - cell size

$I$  - energy integral (table 2)

Energy integral I of H<sub>2</sub>-air mixture (Guirao et al. 1982)

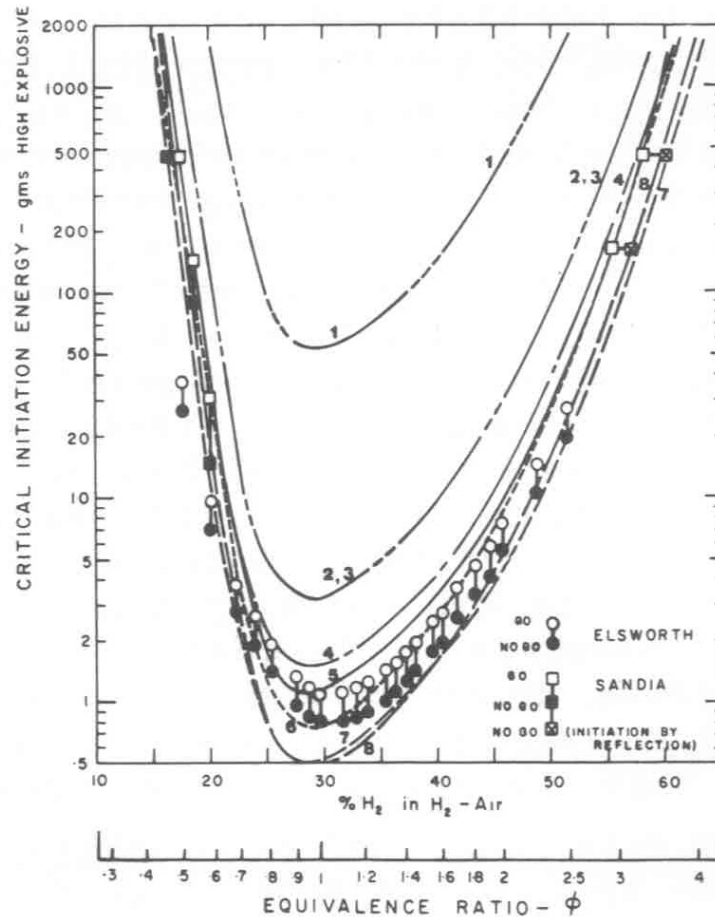
% H <sub>2</sub>	I	% H <sub>2</sub>	I
15	0.64765	45	0.76532
20	0.74232	50	0.71585
25	0.87237	55	0.67299
29.6	0.99306	60	0.63518
30	0.99527	65	0.60108
35	0.91458	70	0.58742
40	0.82589	75	0.54058

Fuel-Air Mixtures ( $T_o = 300\text{ K}$ ,  $P_o = 1\text{ atm}$ )



Model vs. Experimental data

## Models vs. Experimental data



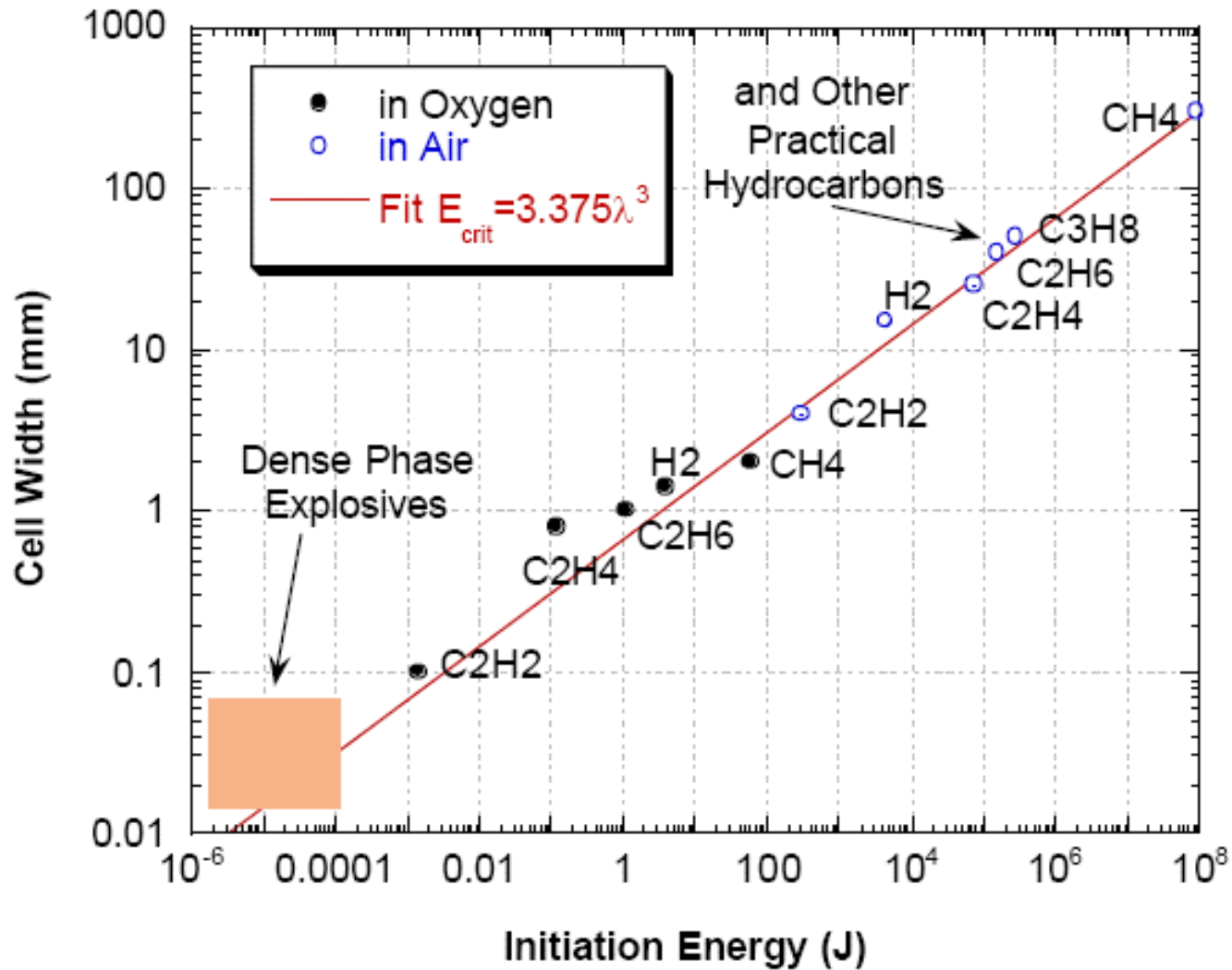
*Theoretical curves:*

- 1 - Edwards (1976a) hydrodynamic thickness model;
- 2 - Lee et al. (1982) blast model; 1
- 3 - Lee et al. (1983) hydrodynamic thickness model;
- 4 - detonation kernel model;
- 5 - surface energy model;
- 6 - cell energy model;
- 7 - chemical energy model;
- 8 - work done model.

*(Benedick et al. 1986)*



# Direct initiation of detonation



(Schauer et al. 2005)

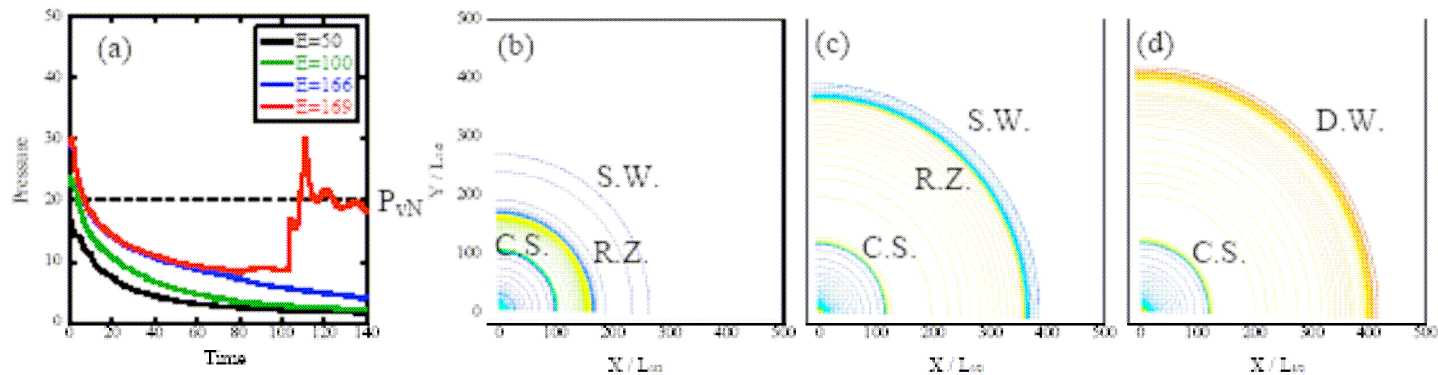


Fig.3 (a) shock pressure histories on the line of  $X=Y$  ( $P_{vN}=20.03$ ). Temperature distributions of (b)  $E=50 \times 10^6$ , (c)  $E=166 \times 10^6$  and (d)  $E=169 \times 10^6$  at  $t=110$  with circular grid. 20 points/ $L_{1/2}$  and 1 point/deg. are set. (D.W. : Detonation Wave, S.W. : Shock wave, R.Z. : Reaction Zone, C.S. : Contact Surface)

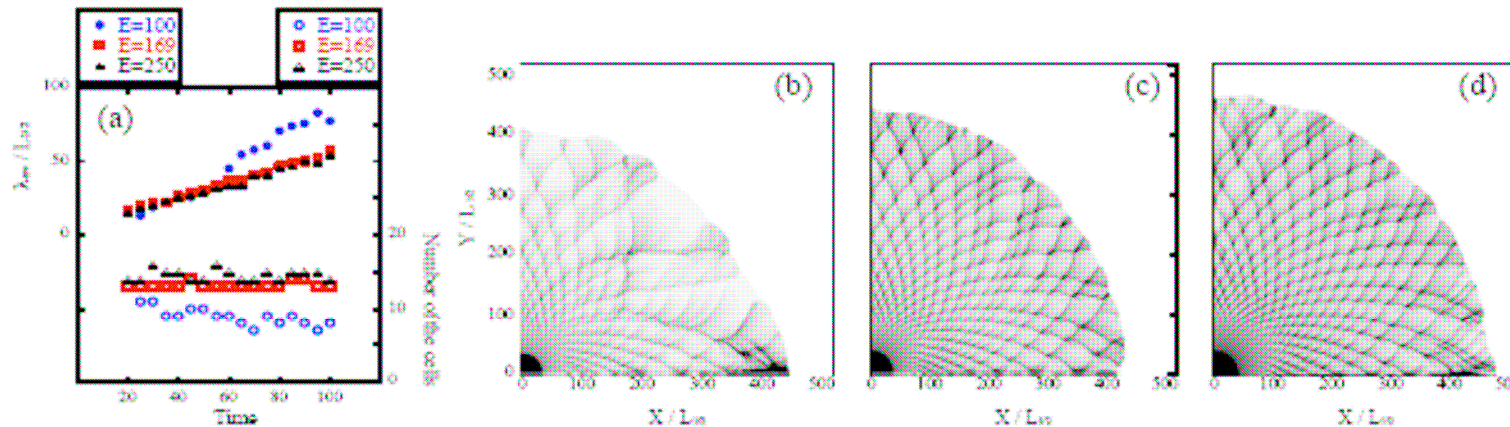
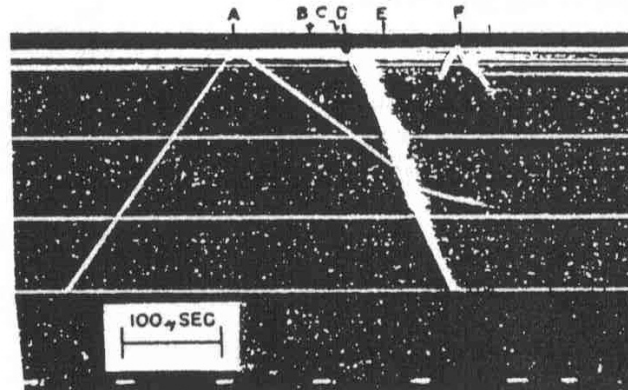
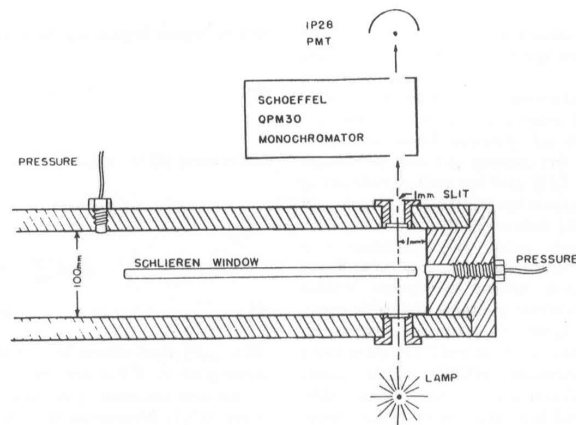
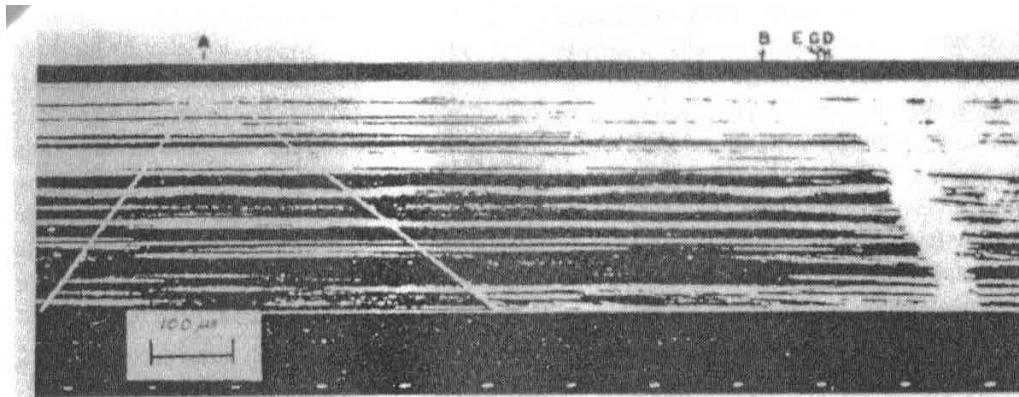


Fig.5 (a) Evolution of the averaged cell width and the cell number. The maximum pressure distributions of (b)  $E=100 \times 10^6$ , (c)  $E=169 \times 10^6$  and (d)  $E=250 \times 10^6$ . 10 points/ $L_{1/2}$  are set in X and Y directions.

Source:  
Nirasawa & Matsuo  
21st ICDERS, July  
23-27, 2007,  
Poitiers

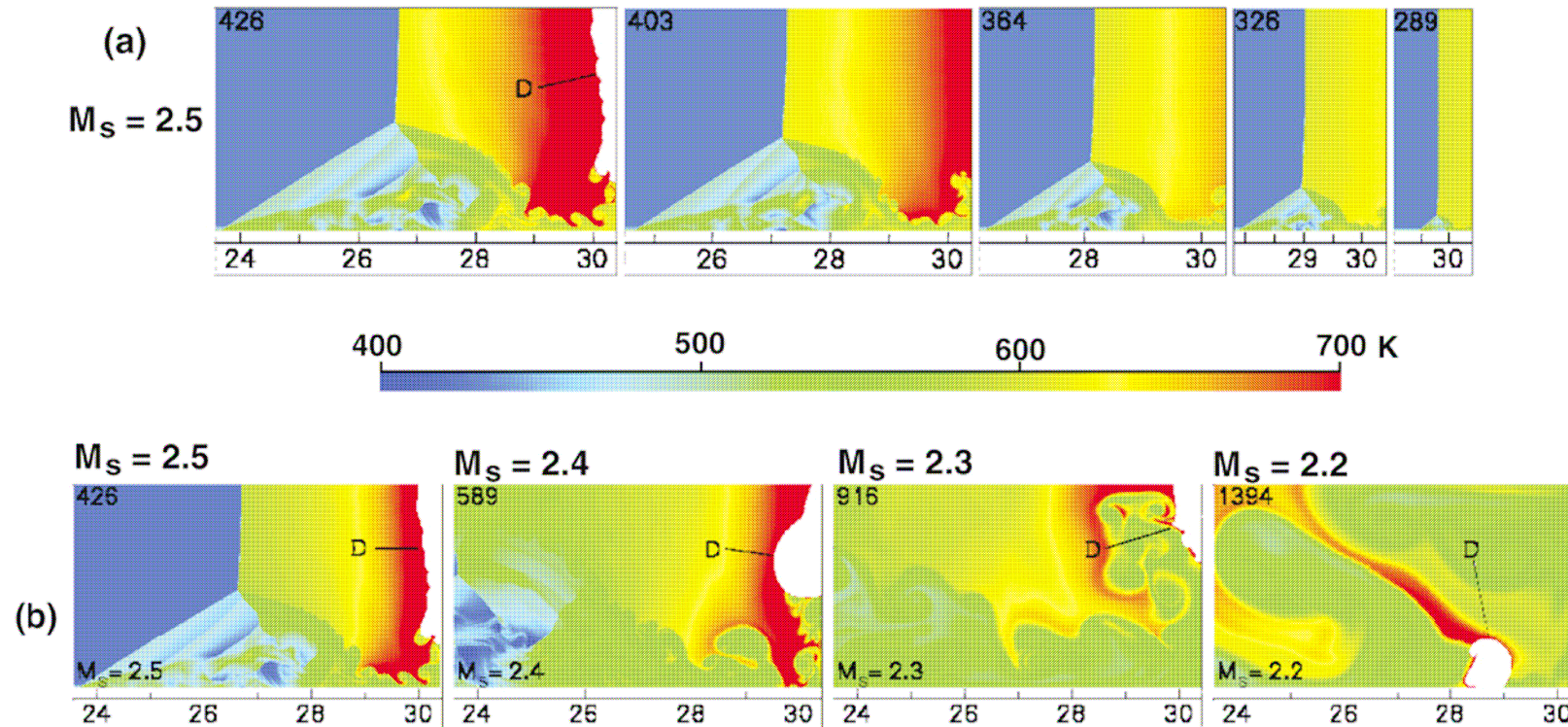


Strong ignition;  
 $2\text{H}_2 + \text{O}_2 + 7\text{Ar}$

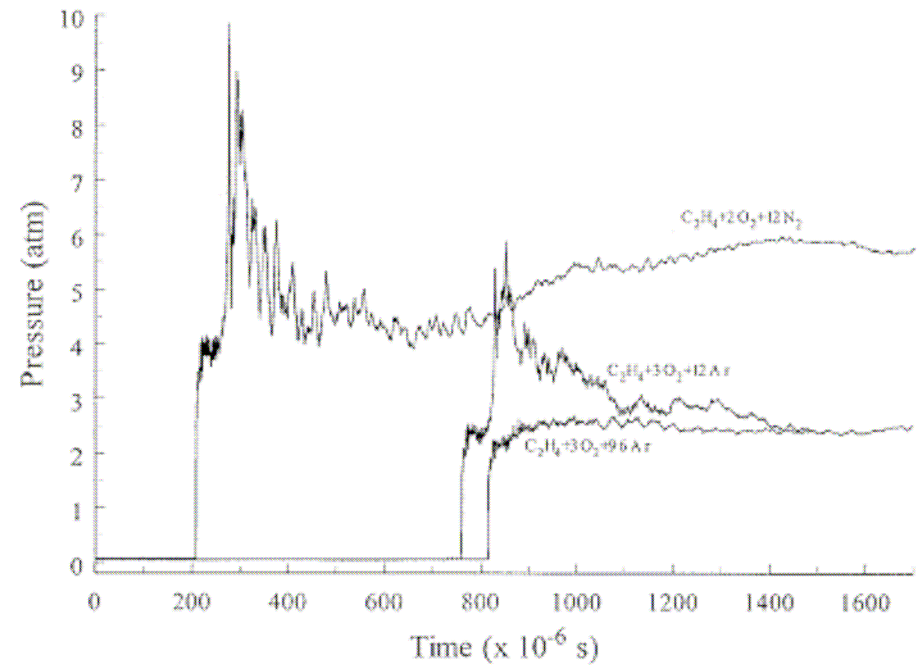
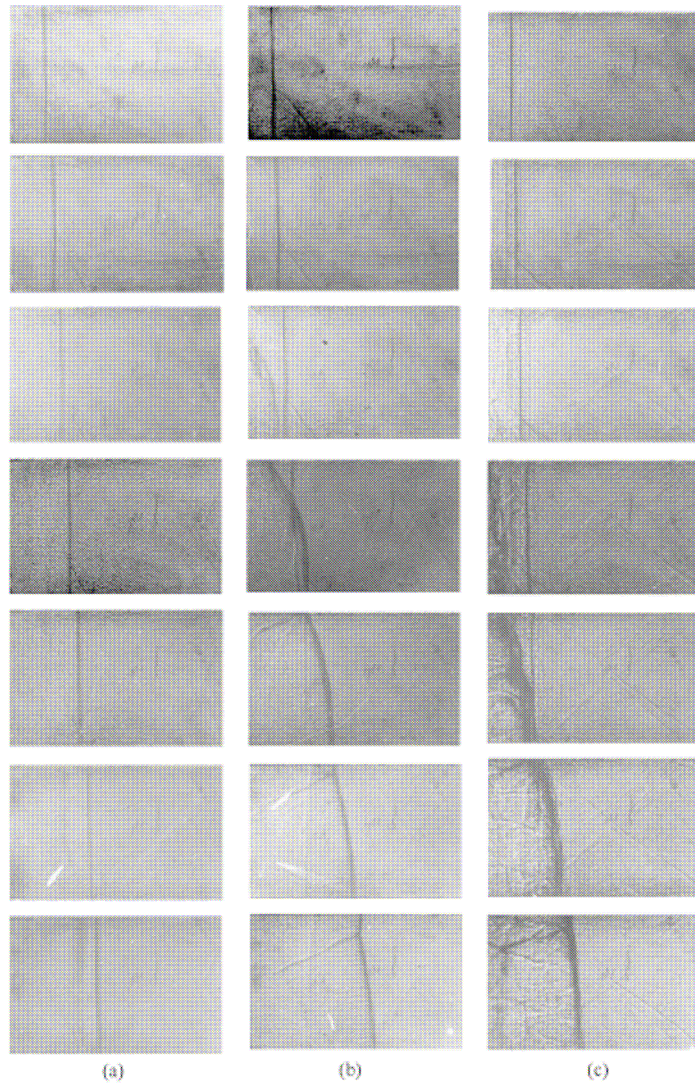


Weak ignition;  
 $8\text{H}_2 + 2\text{O}_2 + 90\text{Ar}$

(Cohen & Larsen, 1967)



(Gamezo et al., 2001)



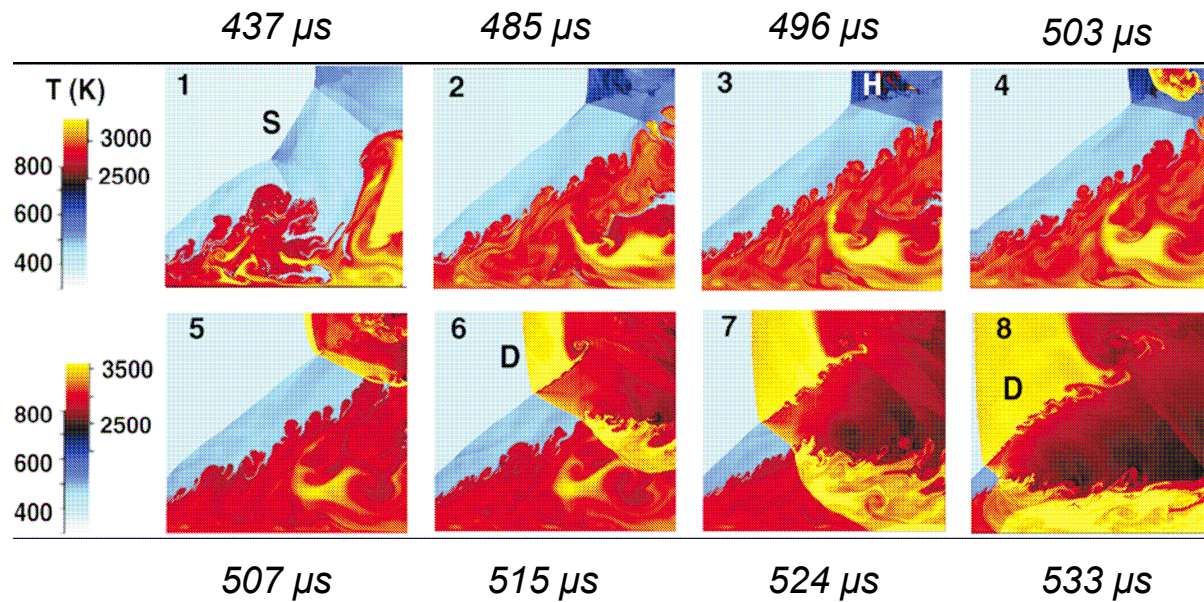
- (a)  $C_2H_4 + 3O_2 + 96Ar$  (incident shock Mach number 2.65),  
 (b)  $C_2H_4 + 3O_2 + 12Ar$  (incident shock Mach number 2.64),  
 (c)  $C_2H_4 + 3O_2 + 12N_2$  (incident shock Mach number 3.11).

Initial pressure 0.0526 bar; 10  $\mu s$  frame spacing.

(Brown & Thomas, 1999)

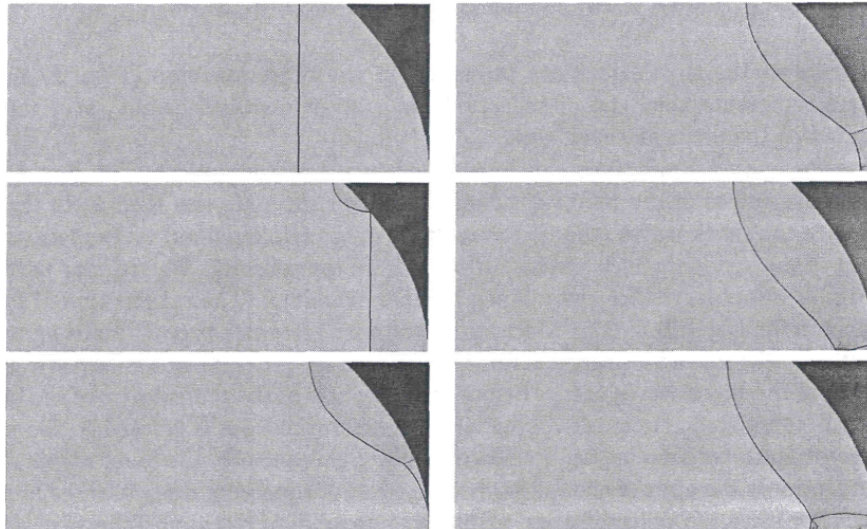
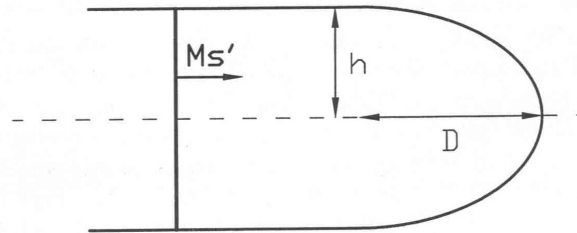
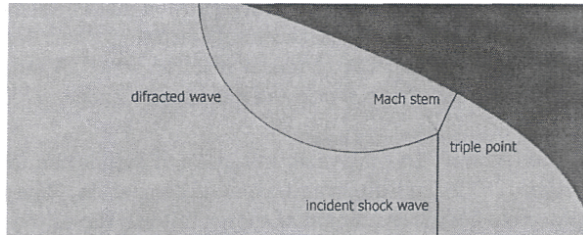
*S* – shock wave      *H* – hot spot

*D* - detonation

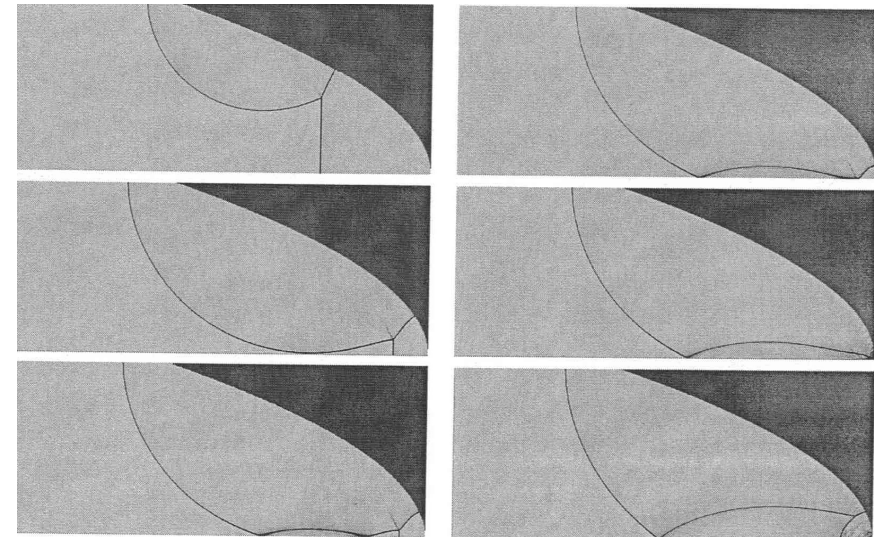


(Gamezo et al. 1978)

# Detonation initiation by shock focusing



(a)

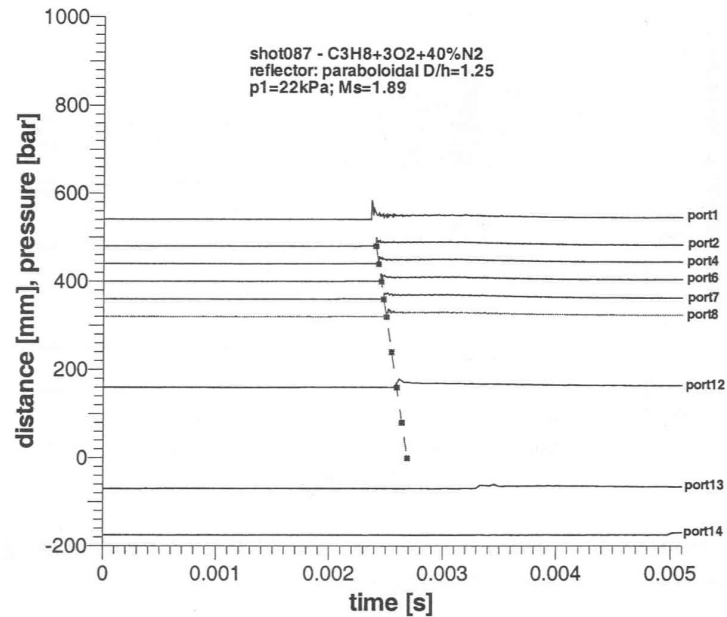


(b)

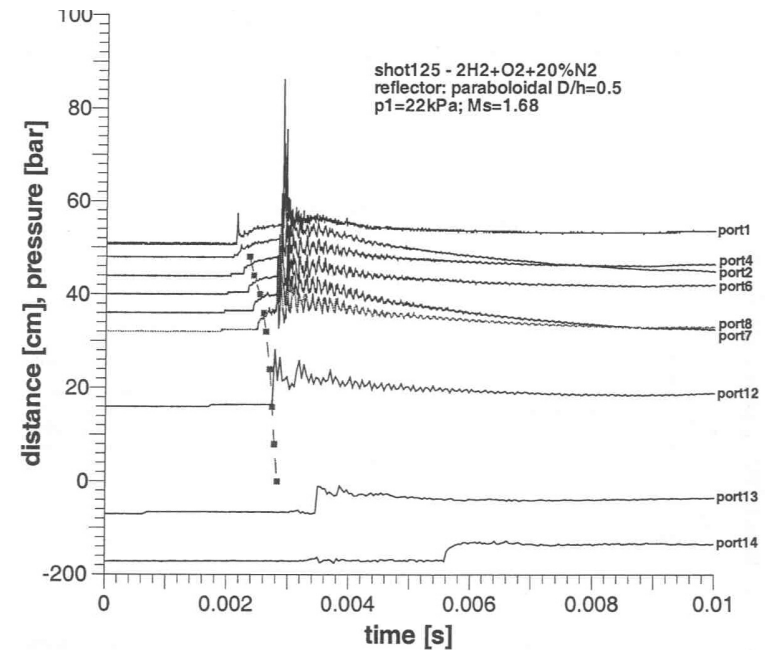
(Buraczewski & Shepherd, 2000)



# Detonation initiation by shock focusing



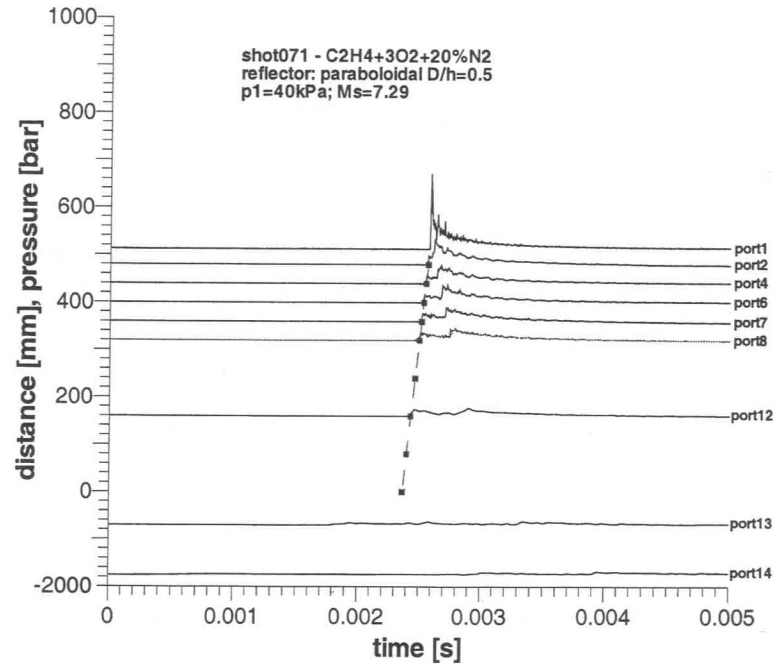
(a) Detonation initiation inside reflector;  
C3H8+5O2+40%N2; Ms = 1.89



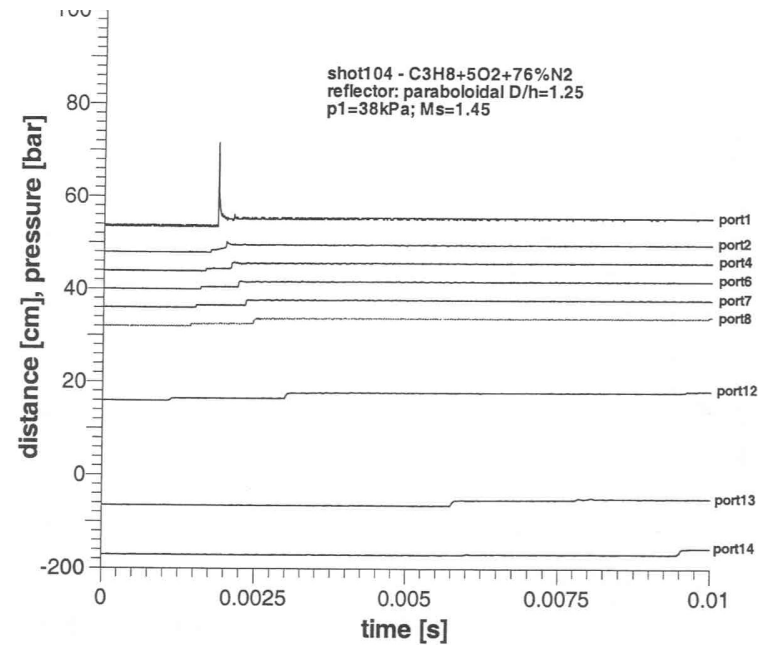
(b) DDT; H2+O2+20%N2; Ms = 1.68

(Buraczewski & Shepherd, 2000)



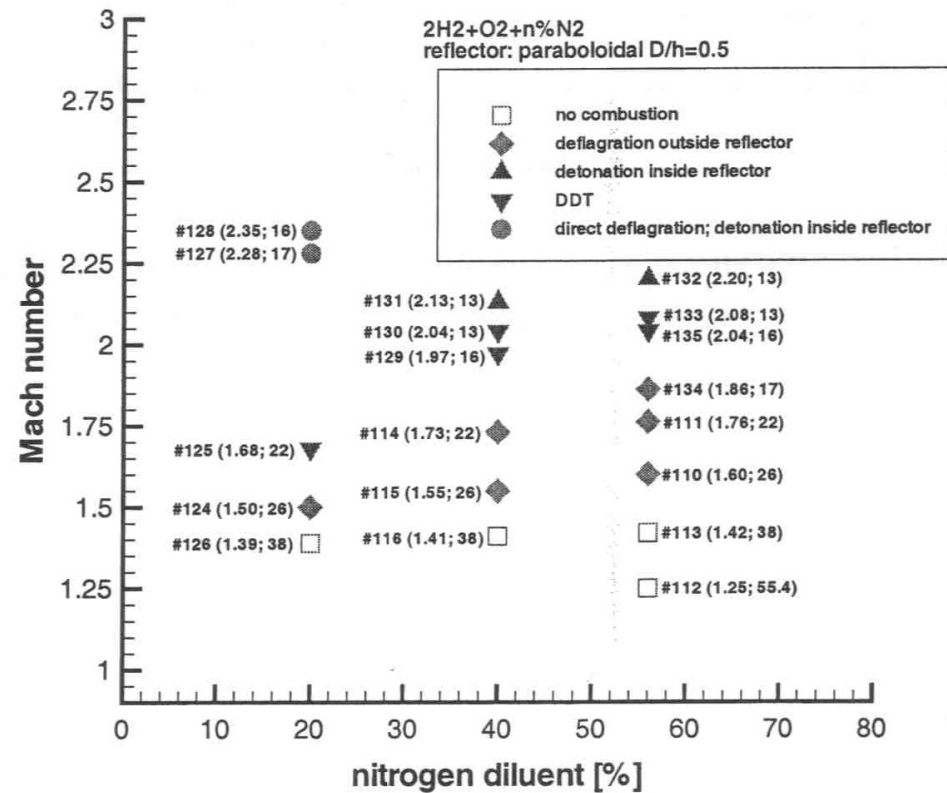


(a) Direct Detonation Initiation;  
 C<sub>2</sub>H<sub>4</sub>+3O<sub>2</sub>+20%N<sub>2</sub>; Ms = 1.41



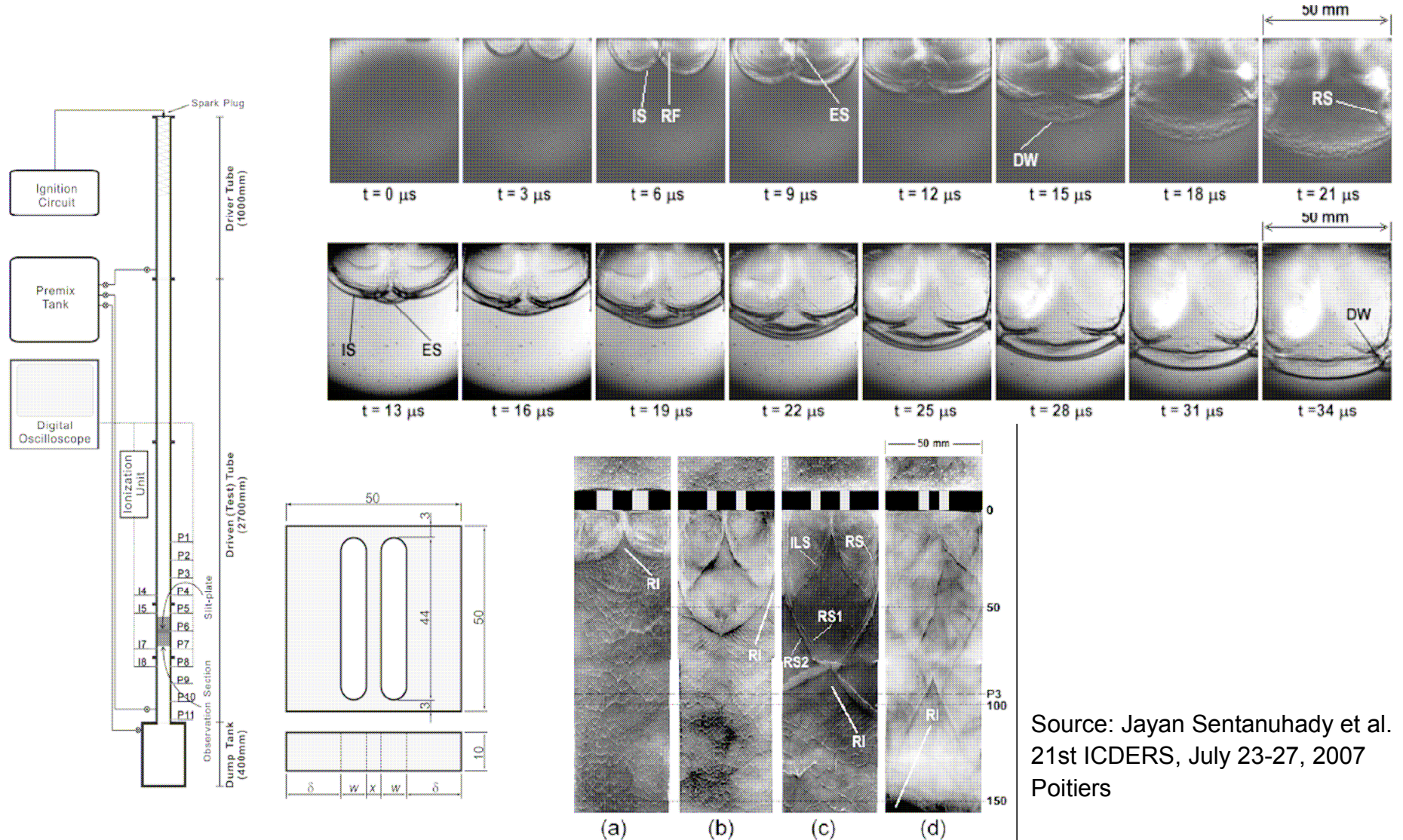
(b) No combustion; C<sub>3</sub>H<sub>8</sub>+5O<sub>2</sub>+76%N<sub>2</sub>; Ms = 1.45

(Buraczewski & Shepherd, 2000)



stoichiometric hydrogen-oxygen mixture  
with varying nitrogen dilution for  
paraboloidal reflector of D/h = 0.5.

(Buraczewski & Shepherd, 2000)



Source: Jayan Sentanuhady et al.  
 21st ICEDERS, July 23-27, 2007  
 Poitiers



# SWACER mechanism



## SWACER – Shock Wave Amplification by Coherent Energy Release

(Lee & Moen, 1980)

- Turbulent mixing  $\Rightarrow$  temperature and concentration gradient – different induction times along the gradient (jet initiation of detonation)
- Ultraviolet irradiation  $\Rightarrow$  gradient of radicals (photochemical initiation)
- Accelerating energy release generates compression waves which are amplified to form a strong shock
- Close coupling between gasdynamics and exothermicity via compression waves



# SWACER mechanism



$D$  – detonation speed

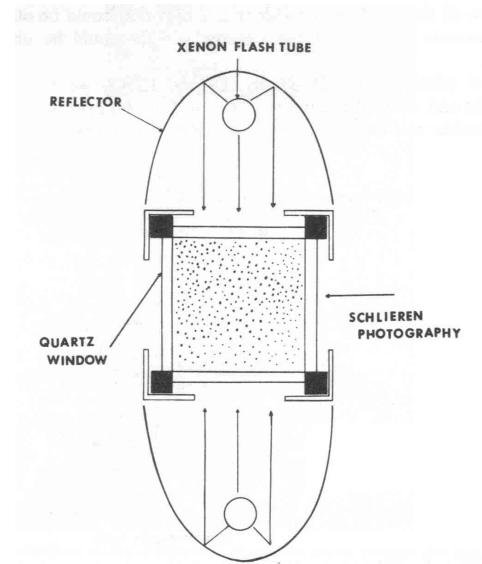
$U_{si}$  – spontaneous ignition wave speed

$a$  – sound speed

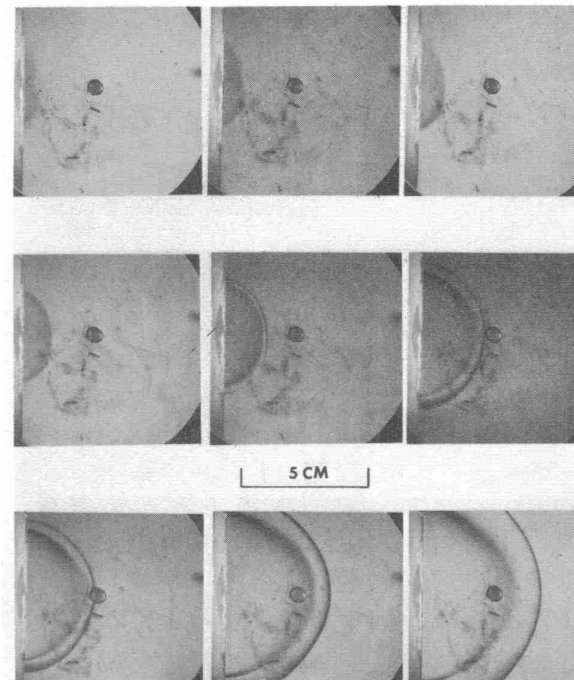
$S_L$  – laminar flame speed

## Zel'dovich (1980) proposed four different regimes:

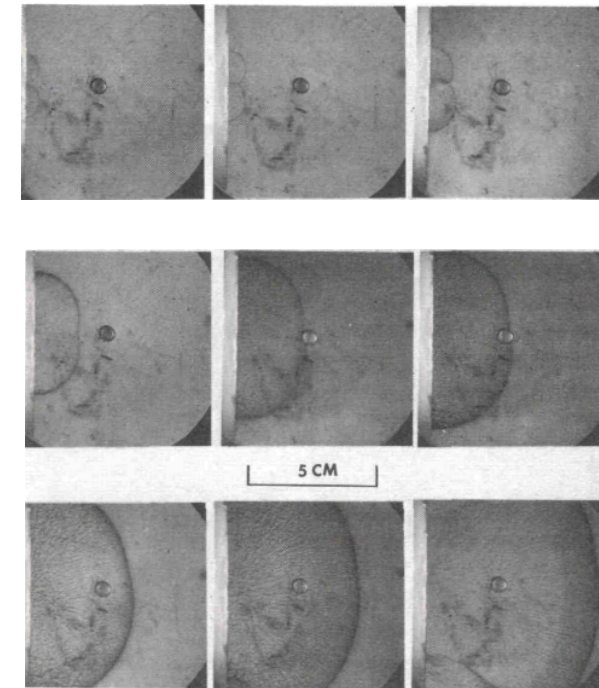
1.  $U_{si} > D$ : reaction wave is so rapid that it resembles a constant volume explosion
2.  $a < U_{si} < D$ : transition to detonation through synchronized initiation
3.  $S_L < U_{si} < a$ : reaction wave propagates with small pressure change across it because the compression wave run away
4.  $U_{si} < S_L$ : diffusion dominates, leading to the formation of laminar flame



$H_2-Cl_2$  at 100 torr

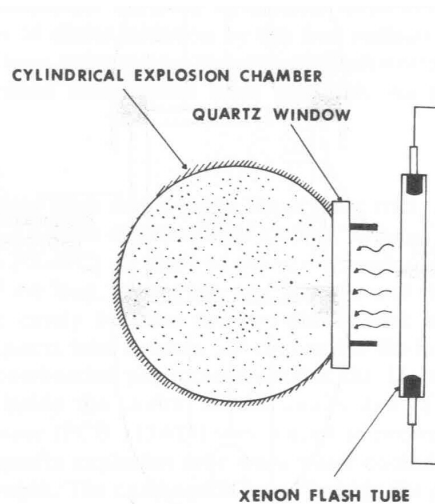


deflagration

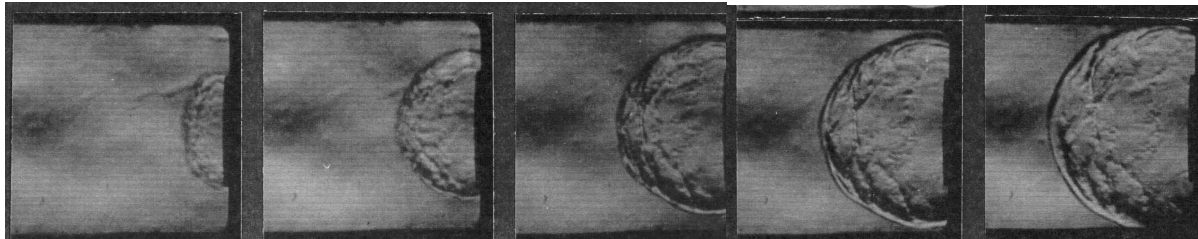


detonation

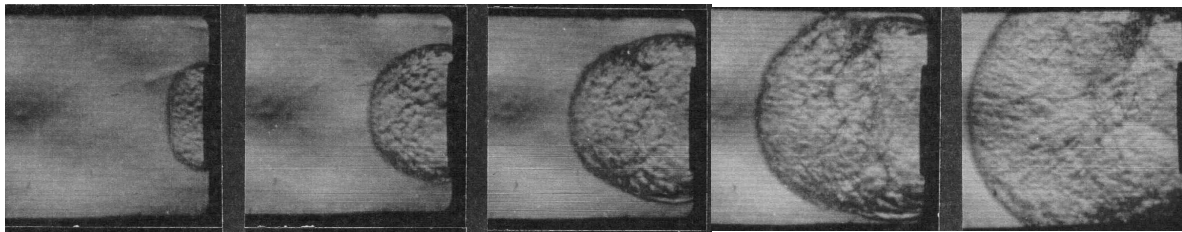
(Lee et al. 1978)



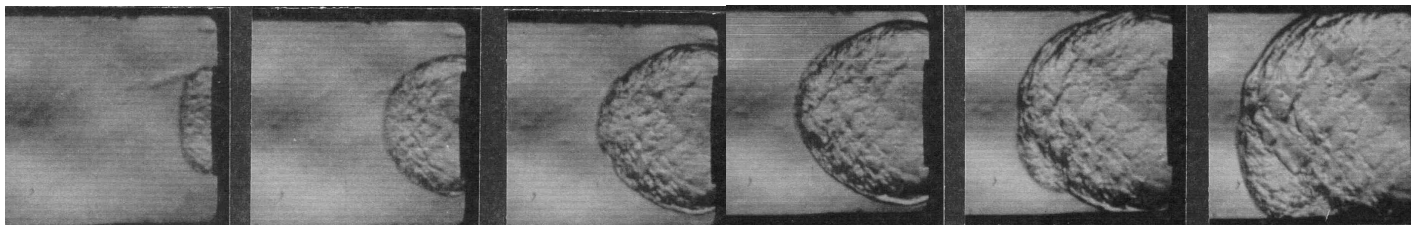
# Jet initiation of detonation



Failure



Transmission

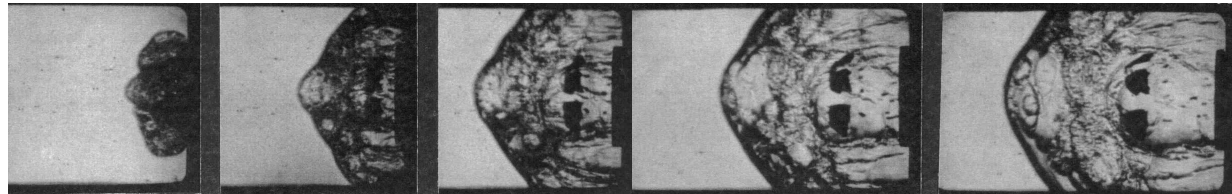


Re-initiation

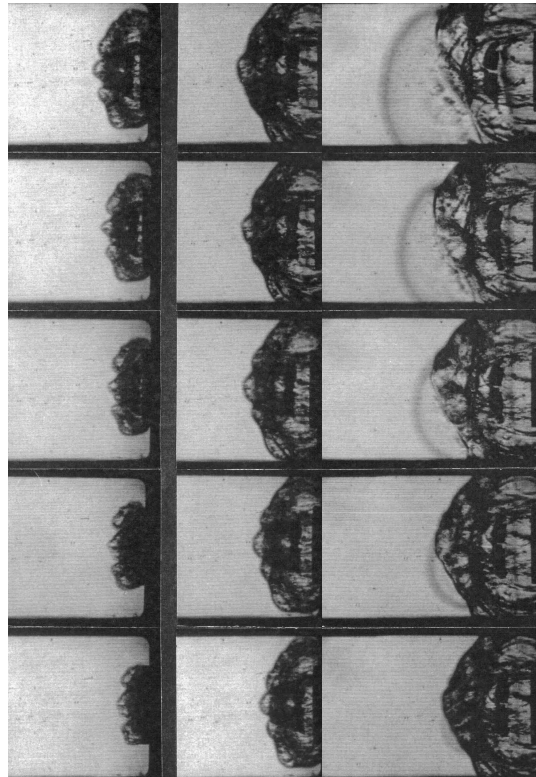
Diffraction of planar detonation into unconfined space

(Inada et al., 1991)

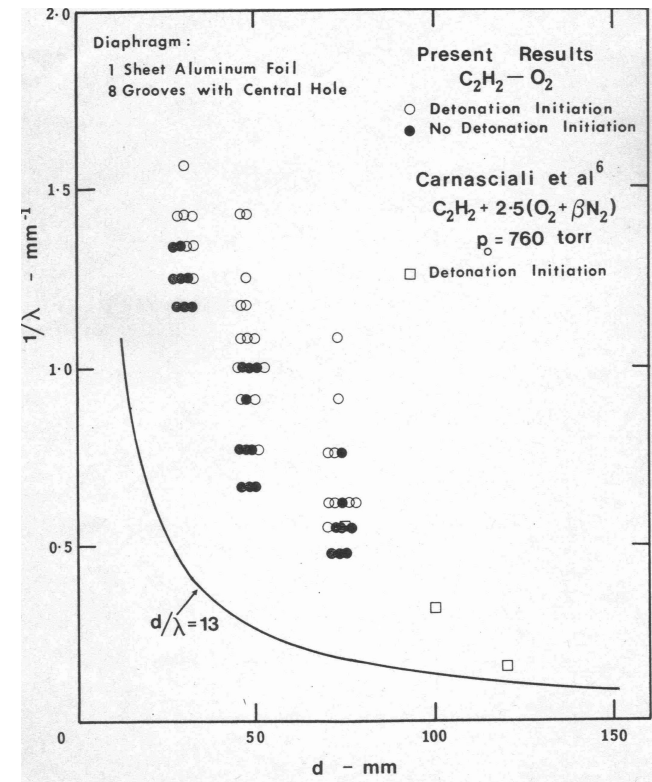
Failure



Initiation



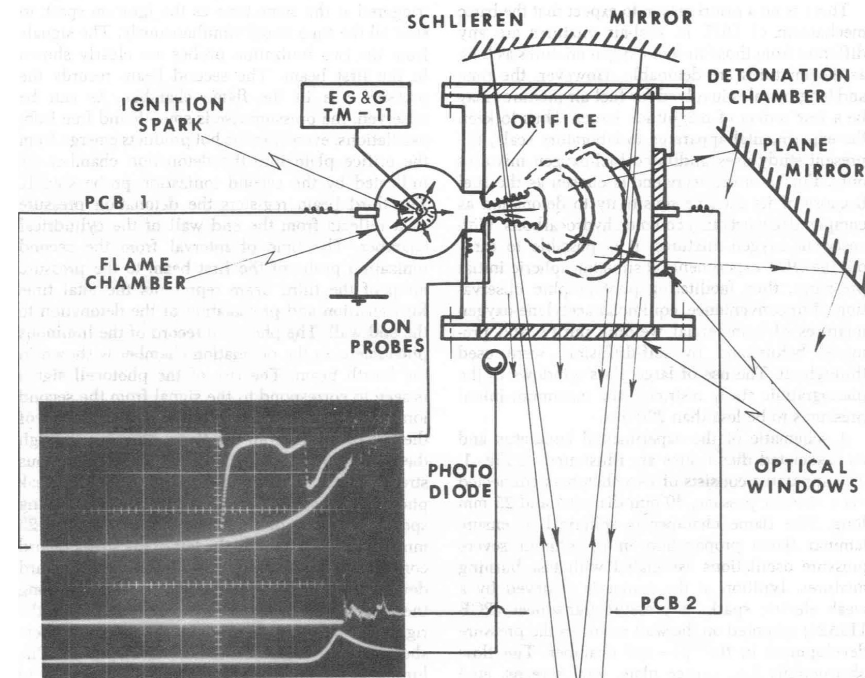
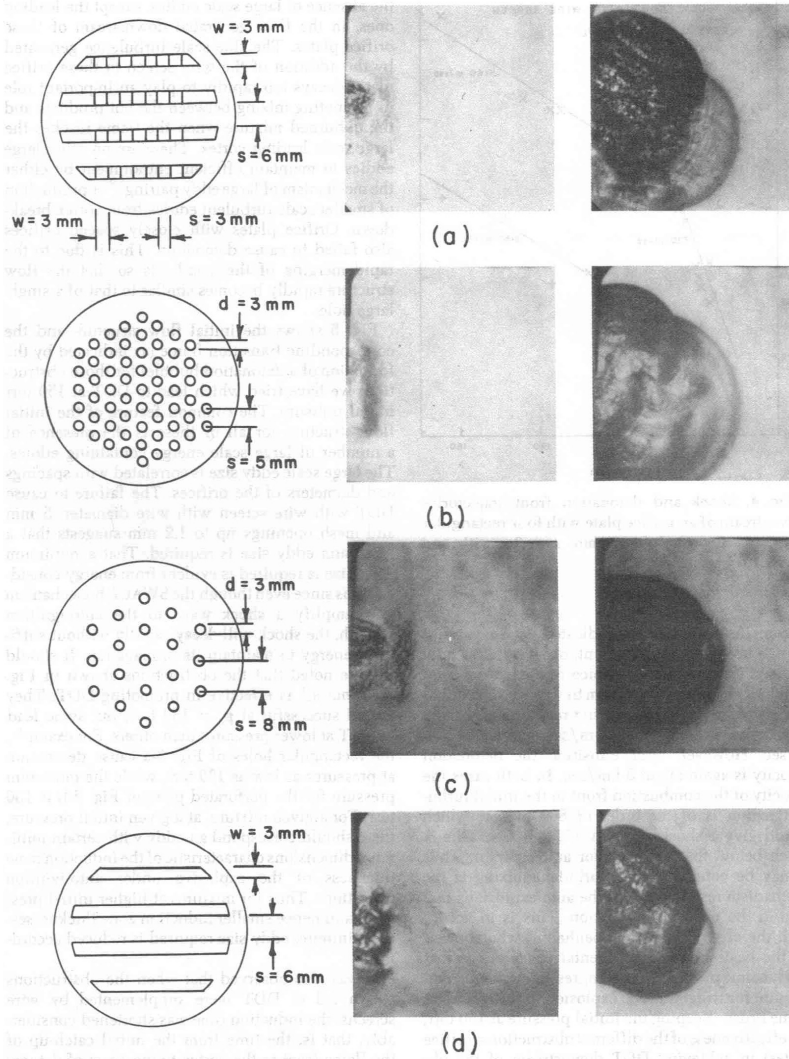
Turbulent  
hot jet



(Inada et al., 1991)

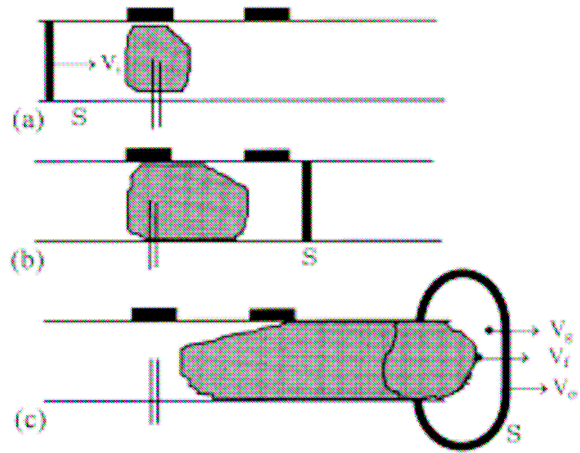


# Jet initiation of detonation



C<sub>2</sub>H<sub>2</sub>-O<sub>2</sub>, 150 torr

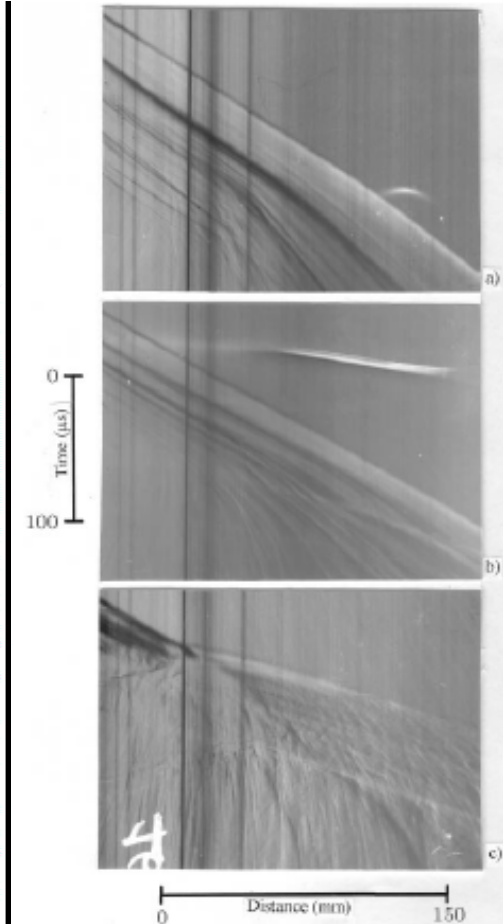
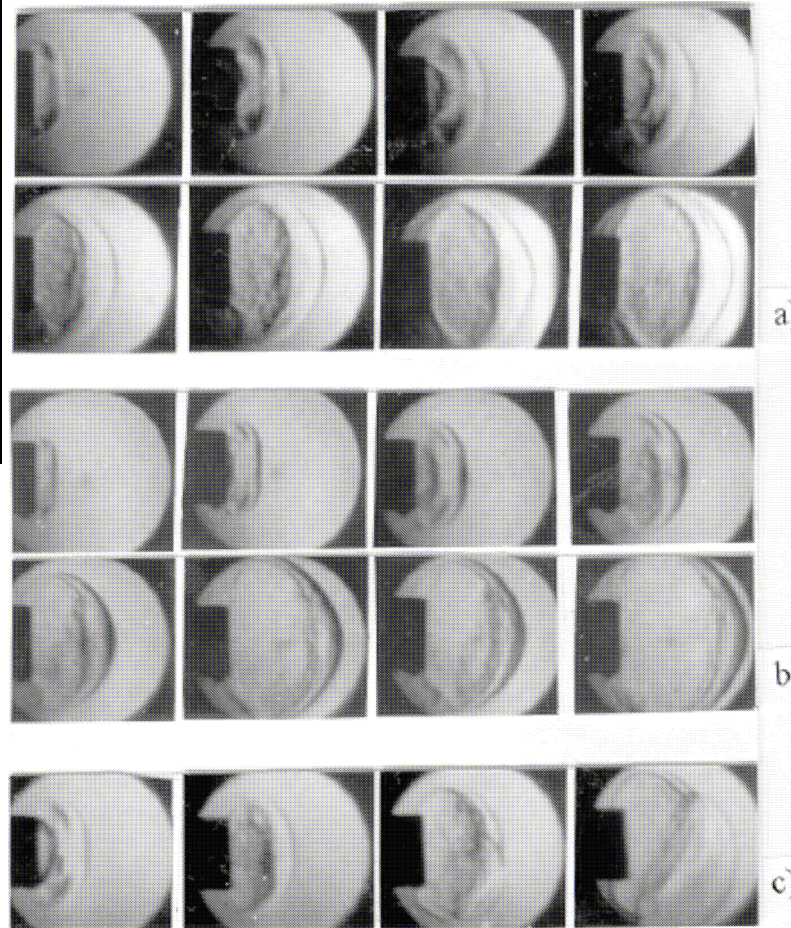
(Knystautas et al., 1978)



C<sub>2</sub>H<sub>4</sub>-O<sub>2</sub>, 60 torr

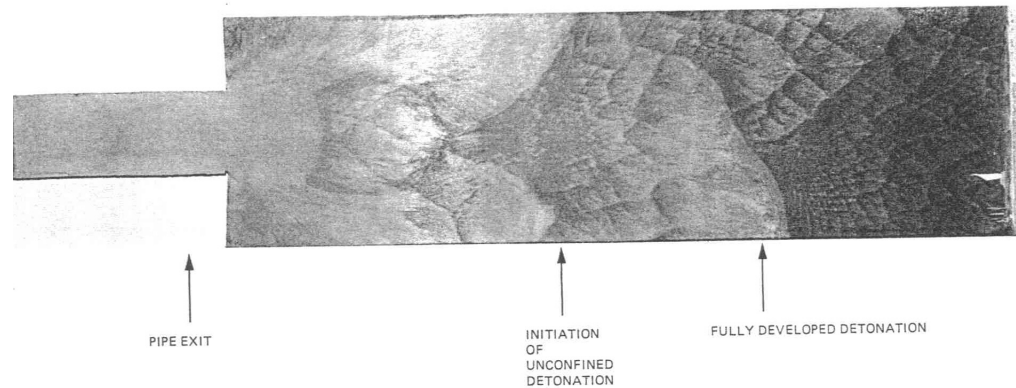
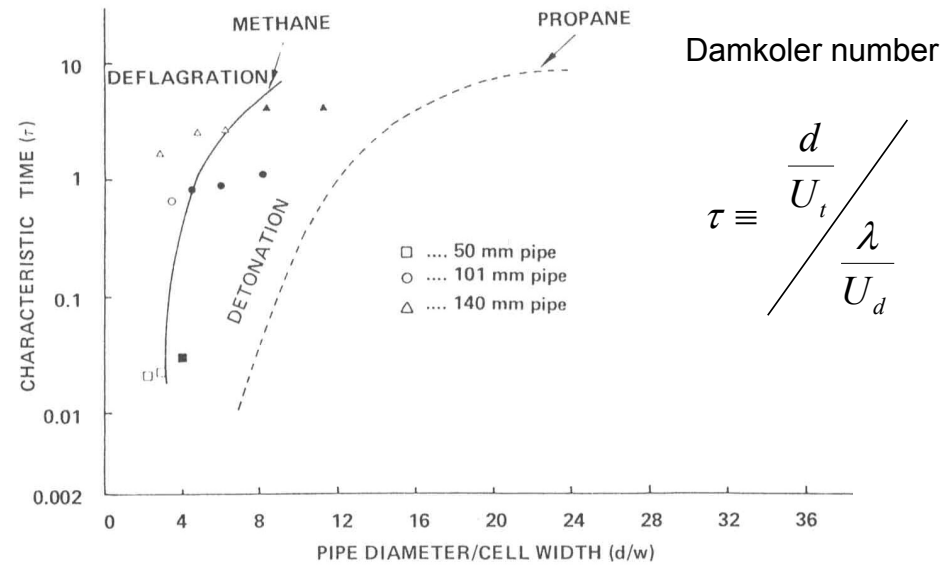
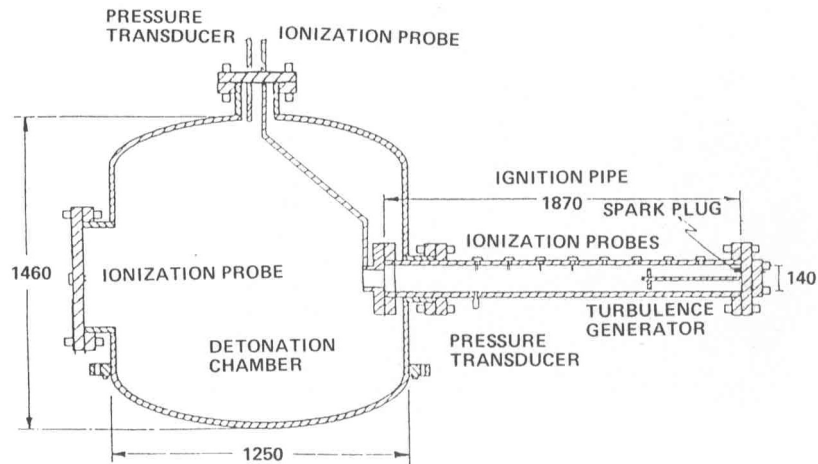
- (a) *failure,*
- (b) *flame acceleration,*
- (c) *transition to detonation.*

$\Delta t = 10 \mu s,$



(Thomas & Jones, 2000)

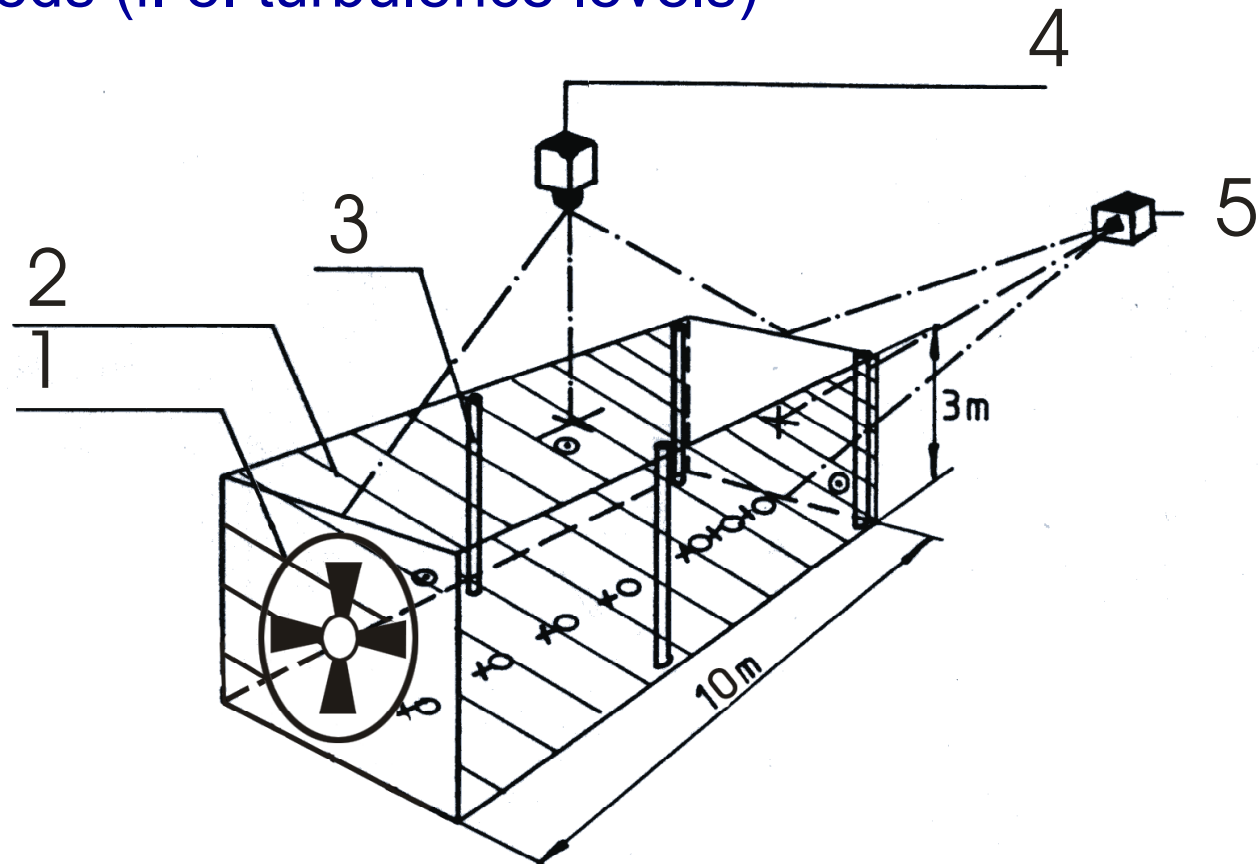
# Jet initiation of detonation



(Ungutt & Shuff, 1988)

Pipe diameter 50 mm for a stoichiometric propane-oxygen with 45.5% nitrogen, 0.4 bar

## Turbulence plus partial confinement Variable fan speeds (i. e. turbulence levels)

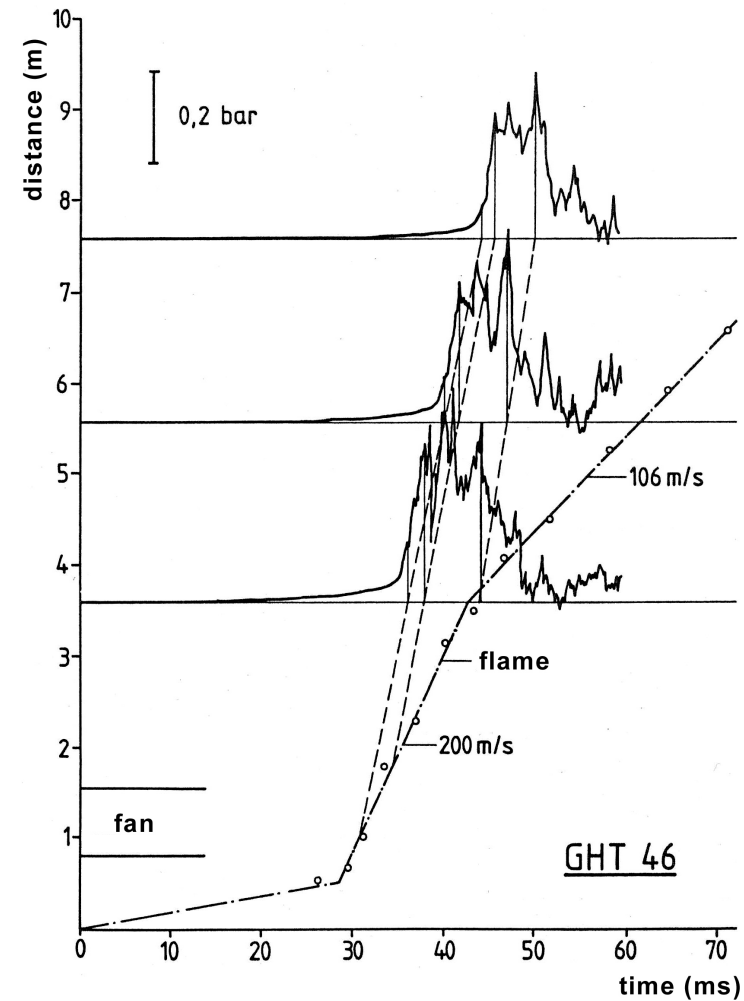


- 1 fan
- 2 partially confined hydrogen/air mixture
- 3 tube
- 4 Hycam- and Locam camera
- 5 Hycam camera
- x, o: pressure transducers

(Courtesy of dr Schneider from Fraunhofer Institute Chemische Technologie)

## Flame Propagation and Pressure/Time-History

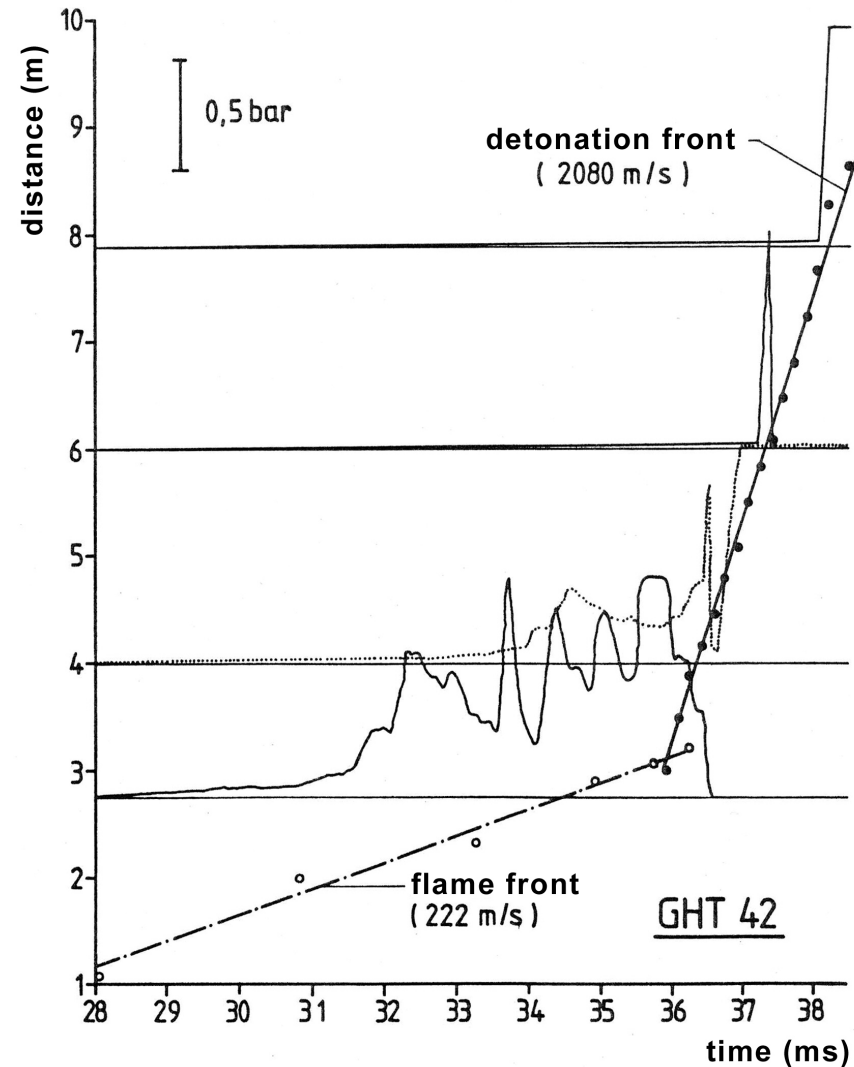
Moderate fan speed  
(i.e. turbulence level)



(Courtesy of dr Schneider from Fraunhofer  
Institute Chemische Technologie)

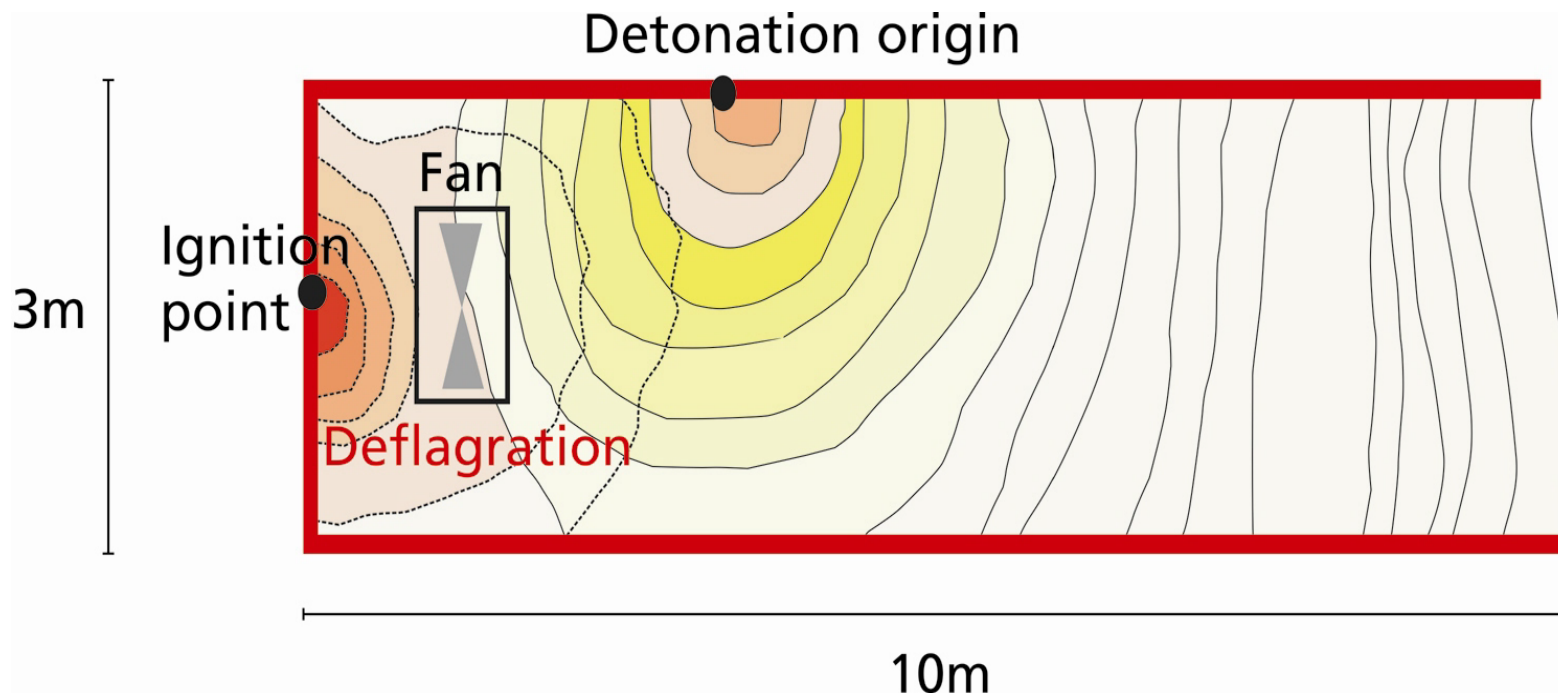
For very high fan speed (i.e. turbulence level):

## Deflagration to Detonation Transition



(Courtesy of dr Schneider from Fraunhofer Institute Chemische Technologie)

Detonation of a hydrogen/air mixture within a "lane", simulated by two parallel walls (top view)



(Courtesy of dr Schneider from Fraunhofer  
Institute Chemische Technologie)

Detonation of a hydrogen/air mixture within a “lane”, simulated by 2 parallel walls (top view)



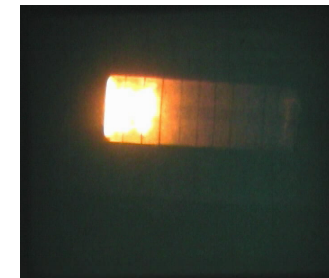
12 ms



30 ms



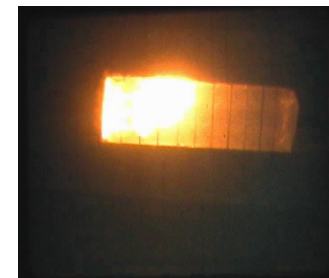
18 ms



36 ms



24 ms



42 ms

(Courtesy of dr Schneider from Fraunhofer Institute Chemische Technologie)





# SWACER in large scale?



Detonation of a hydrogen/air mixture within a "lane", simulated by 2 parallel walls (top view)



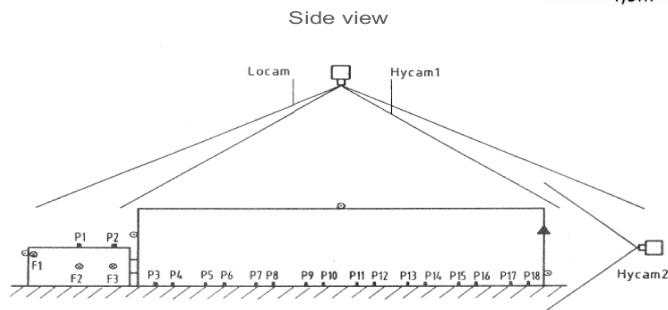
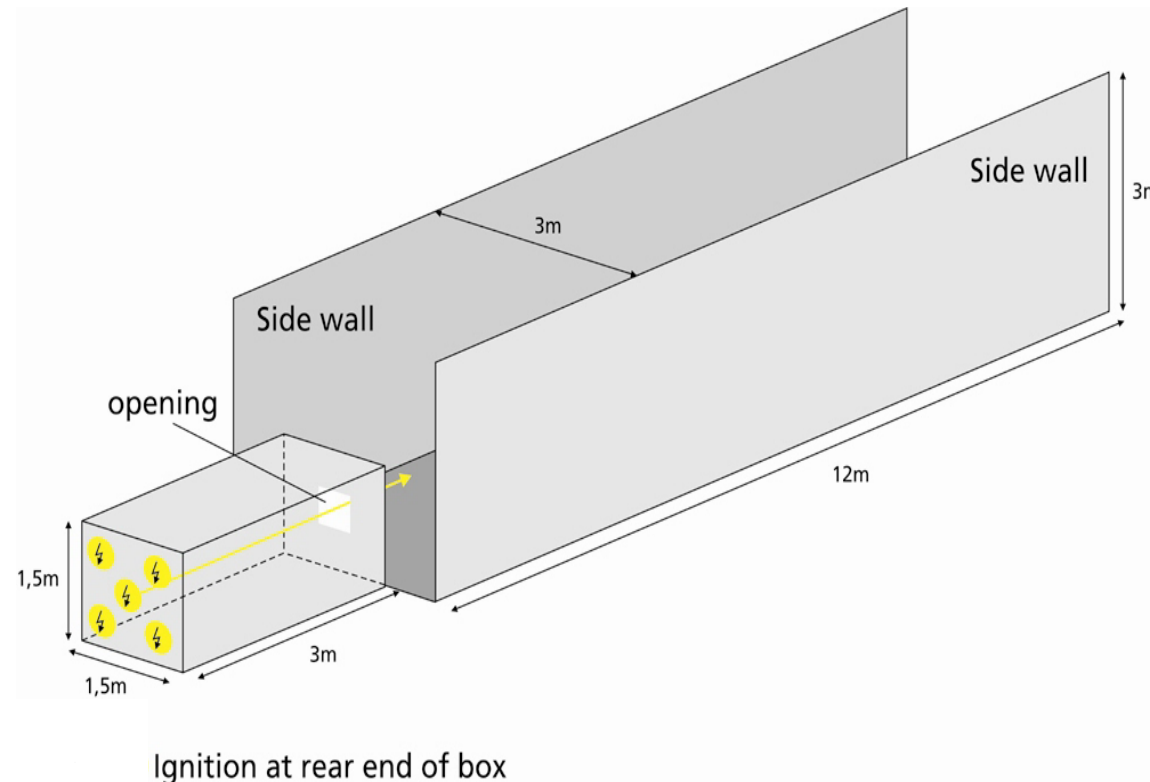
(Courtesy of dr Schneider from Fraunhofer Institute Chemische Technologie)

## Test Site after Detonation



(Courtesy of dr Schneider from Fraunhofer  
Institute Chemische Technologie)

Test setup for igniting a hydrogen/air mixture by means of a flame jet within a “lane”, simulated by two parallel walls



(Courtesy of dr Schneider from Fraunhofer Institute Chemische Technologie)

Flame jet ignition of hydrogen air cloud within a “lane” with subsequent transition to detonation near the ground.



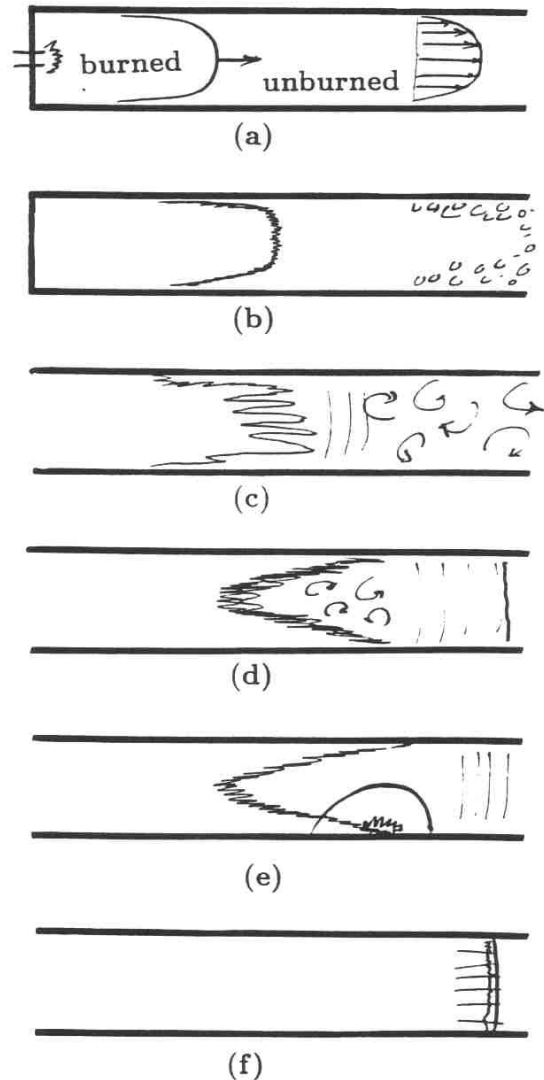
(Courtesy of dr Schneider from Fraunhofer Institute Chemische Technologie)



# Jet initiation of detonation in large scale

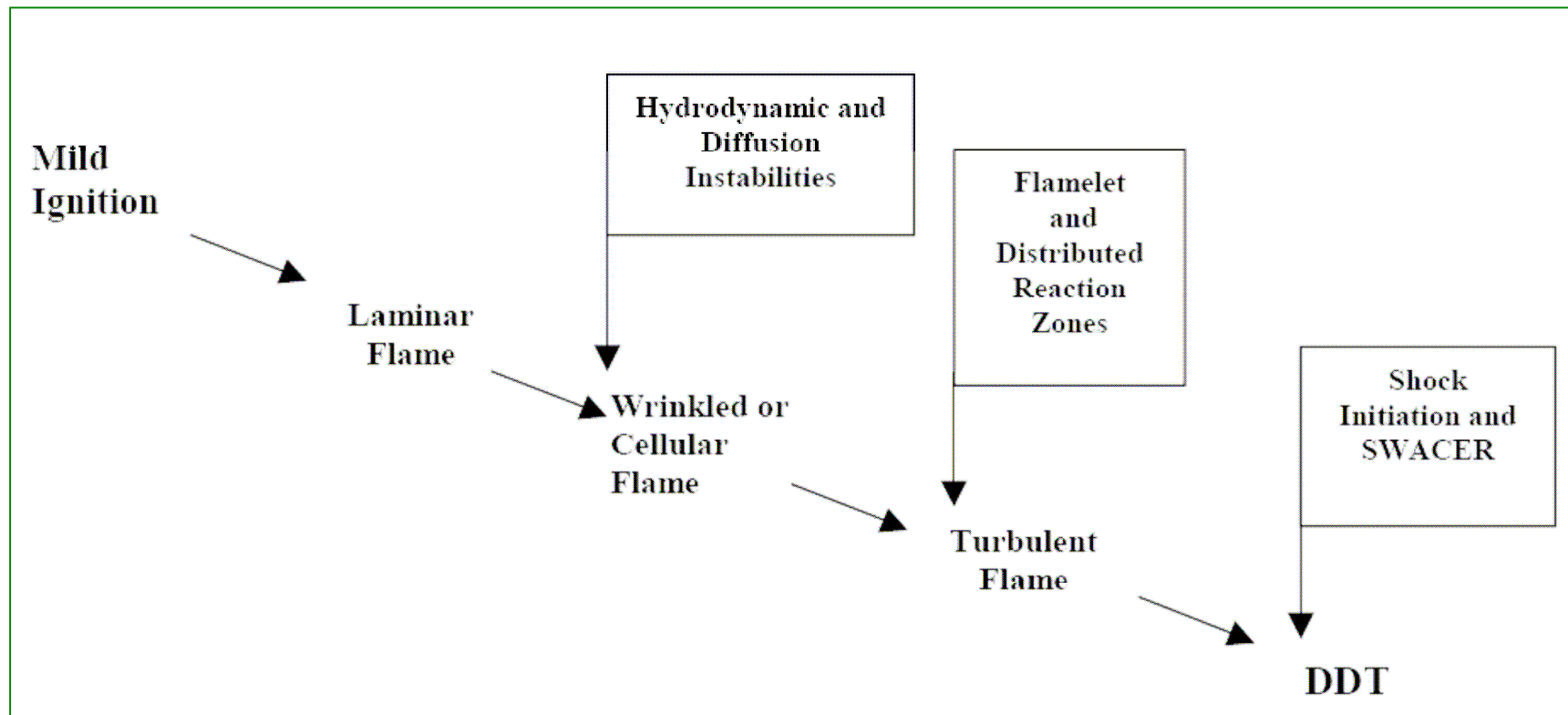


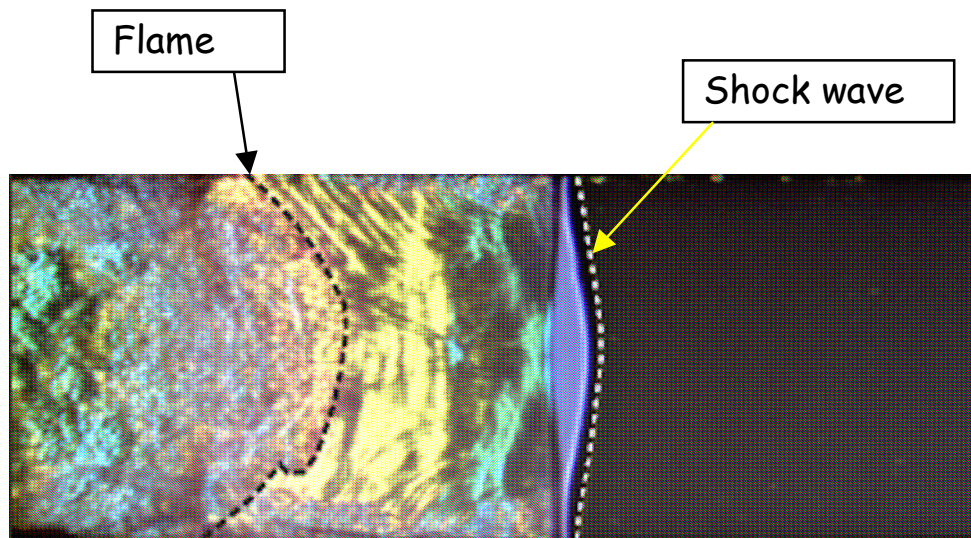
(Courtesy of dr  
Schneider from  
Fraunhofer  
Institute  
Chemische  
Technologie)



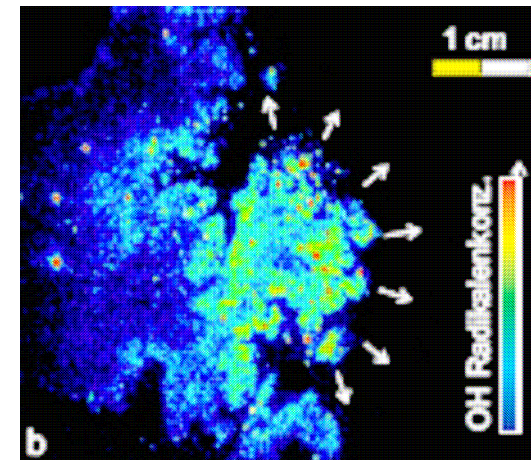
- the initial configuration showing a smooth flame and the laminar flow ahead;
- first wrinkling of flame and instability of the upstream flow;
- breakdown into turbulent flow and a corrugated flame;
- production of pressure waves ahead of the turbulent flame;
- local explosion of a vertical structure within the flame;
- transition to detonation.

(Shepherd&Lee, 1992)





Schlieren image of a fast deflagration wave (22% H<sub>2</sub> in air), flame velocity 1200 m/s;



OH radical distribution of a fast deflagration wave, flame velocity 850 m/s, 17,5% H<sub>2</sub> in air;

*(Eder, 2001)*



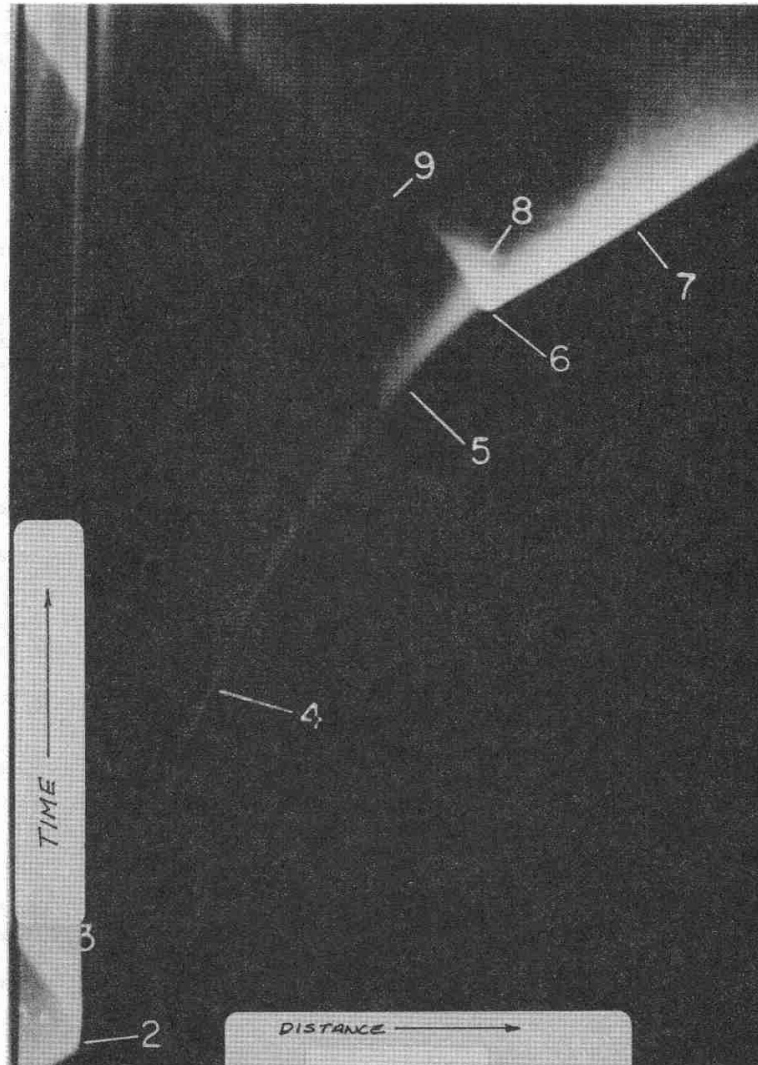


# Transition distance to DDT



Depends on:

- Combustible mixture (chemistry and thermodynamics)
- Tube diameter - for hydrogen-air in smooth tube:
  - 8 m in 50 mm tube
  - 30 m in 400 mm tube
- Ignition source
- Obstacles, wall roughness
- Initial conditions
- ???



## Streak direct photograph

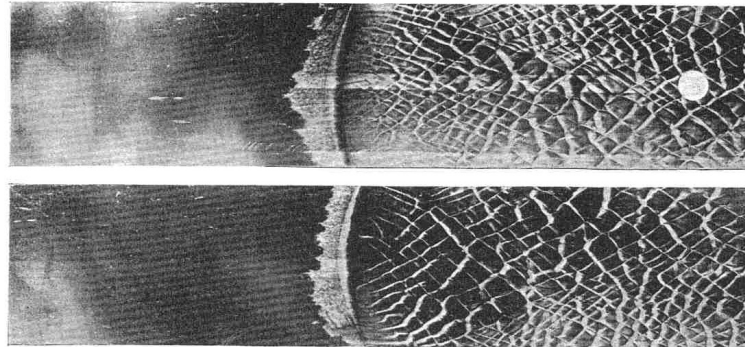
4, 5 - accelerating flame

6 - explosion ahead of the flame

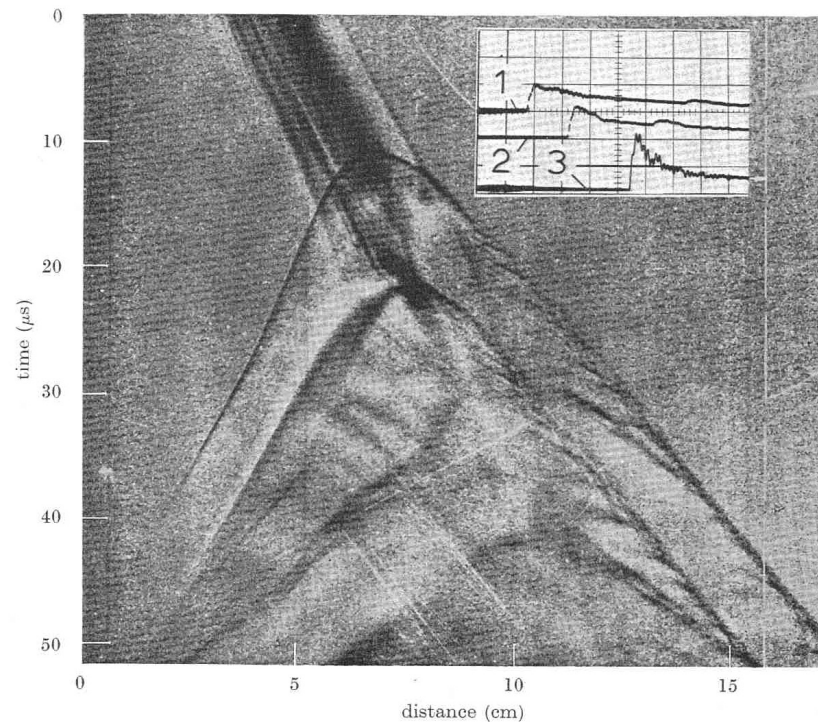
7 - detonation

8, 9 - retonation wave

(Lee, 1978)

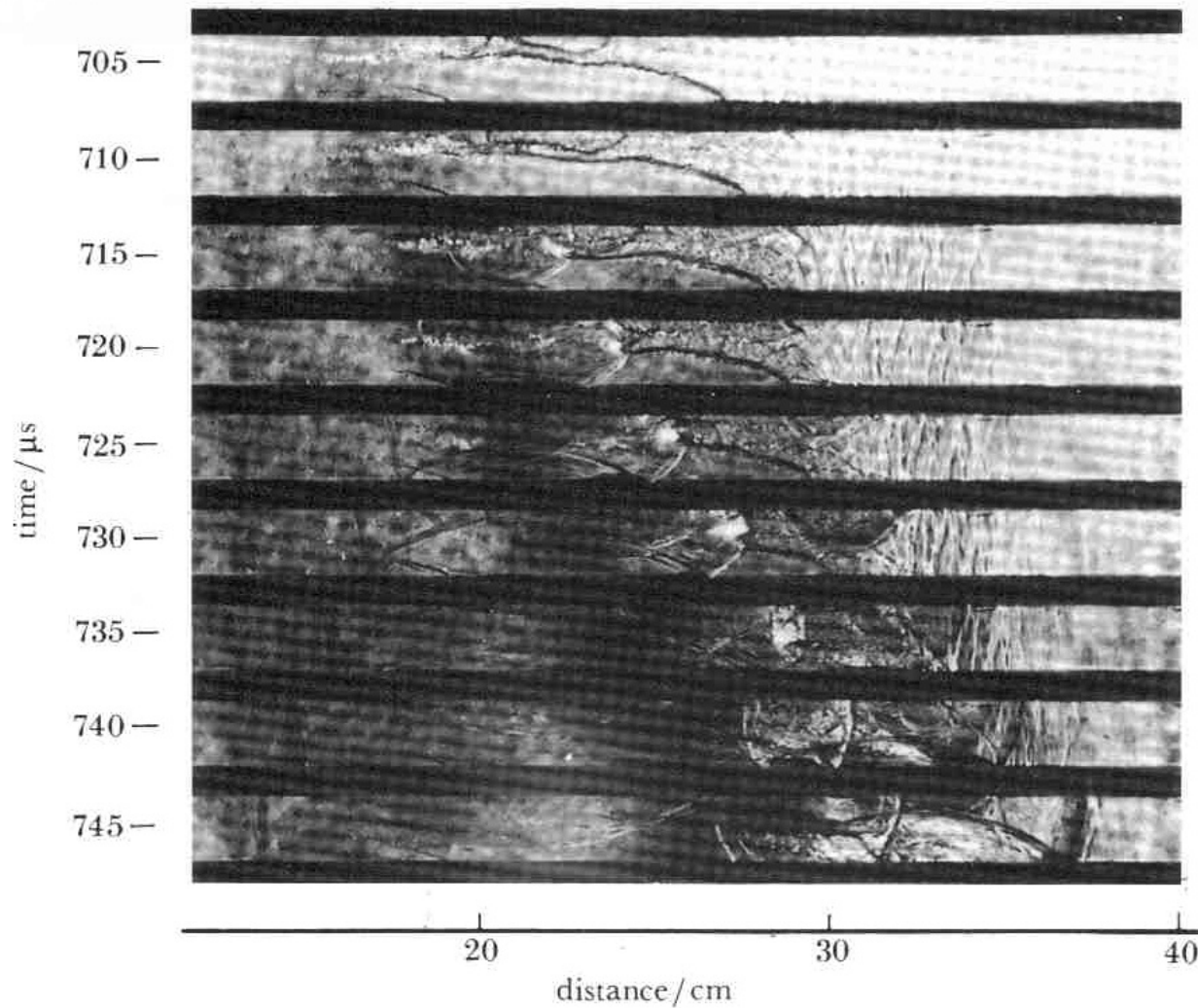


Streak schlieren photograph



(Myer&Oppenheim, 1965)

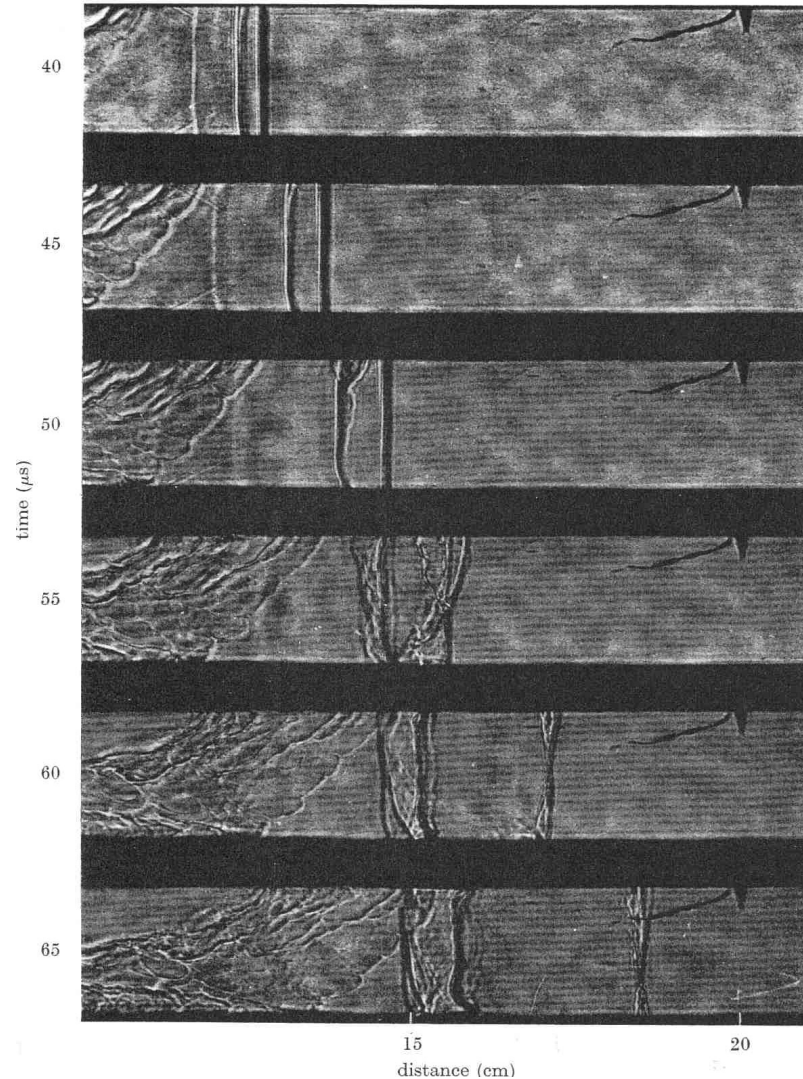
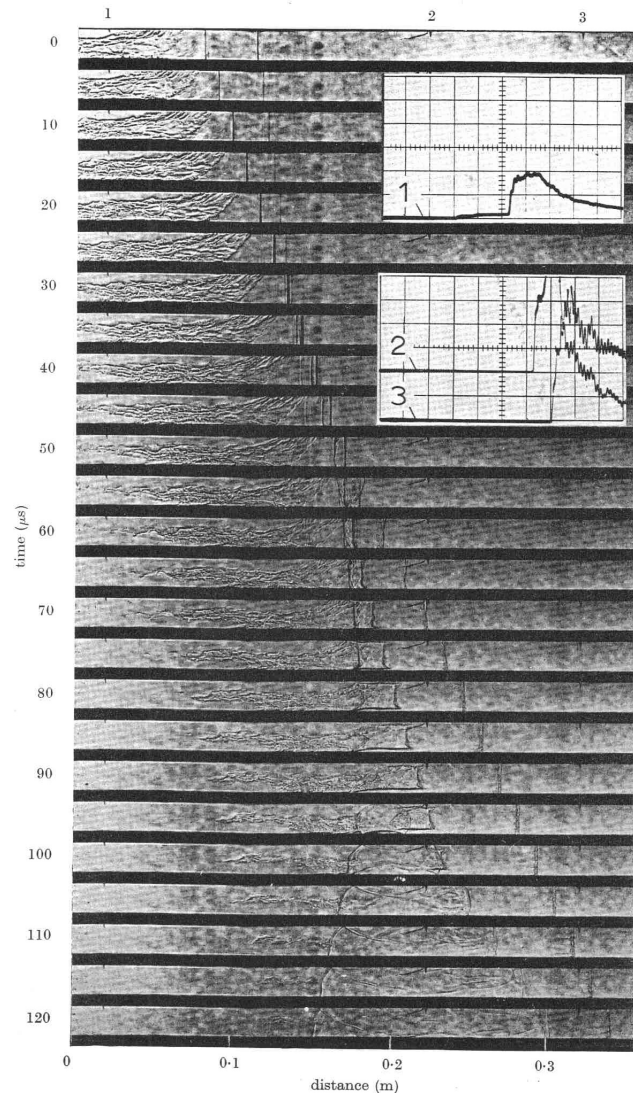
# DDT in smooth tube



Schlieren framing  
photographs by  
rotating mirror  
camera

*(Myer&Oppenheim, 1965)*

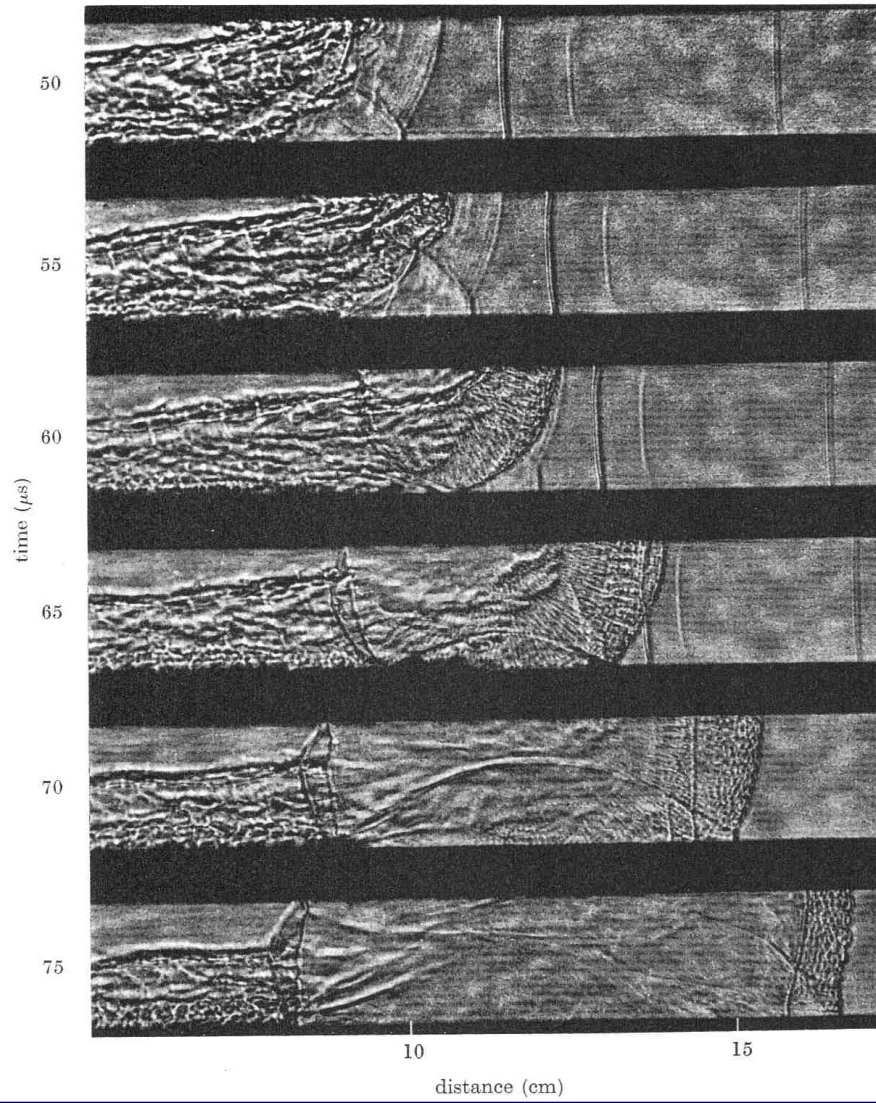
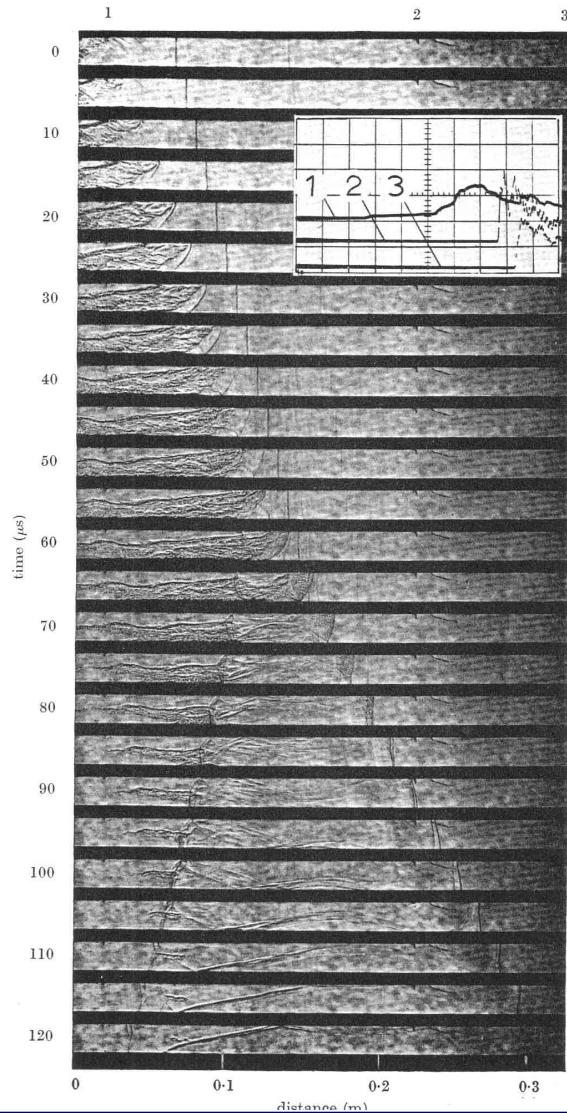
# DDT in smooth tube



Schlieren framing  
photographs  
by rotating  
mirror camera

(Urtiev&Oppenheim,  
1965)

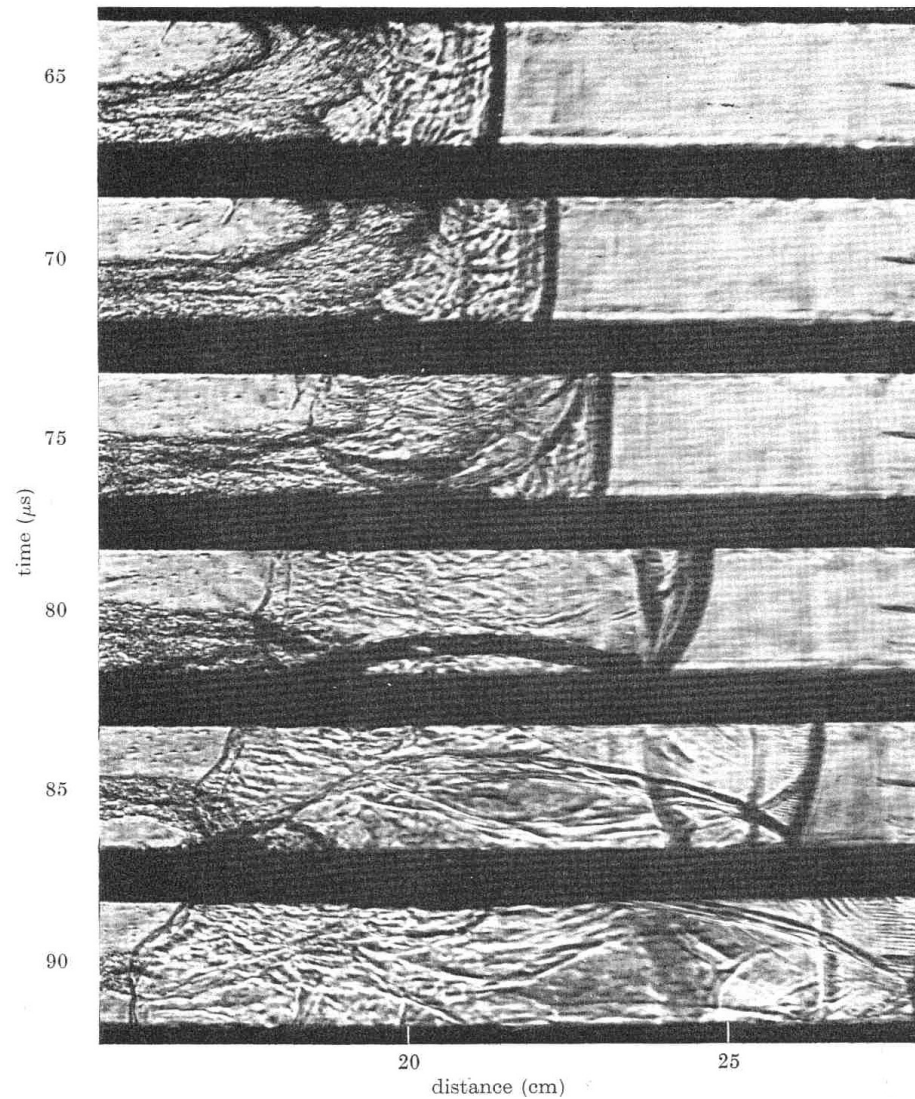
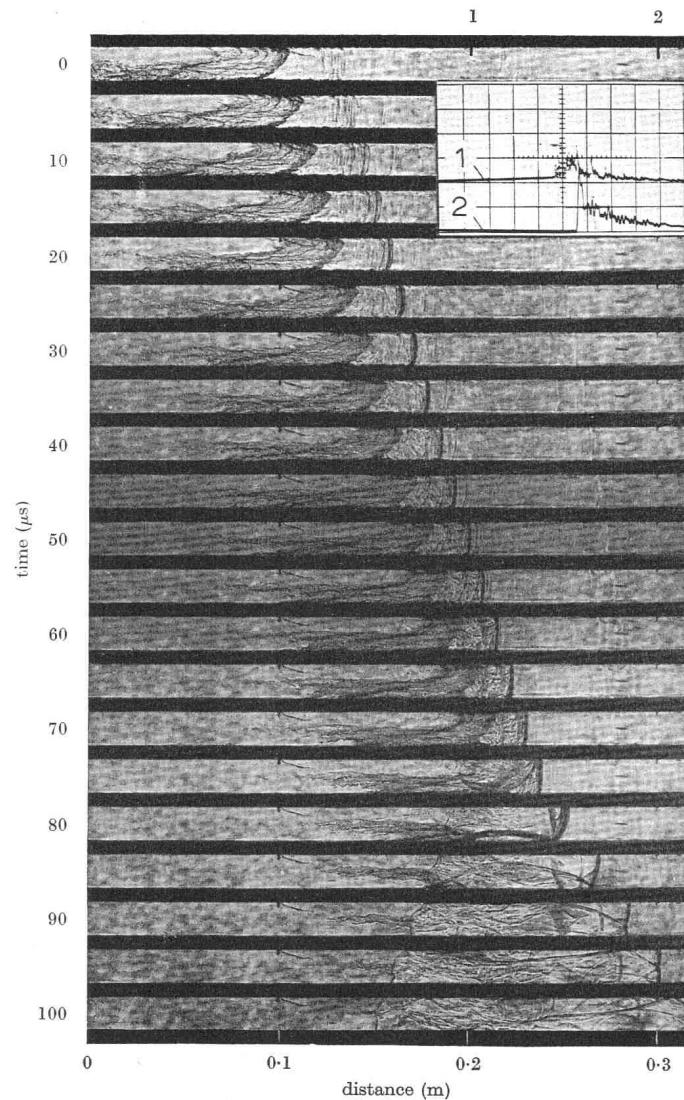
# DDT in smooth tube



Schlieren framing  
photographs  
by rotating  
mirror camera

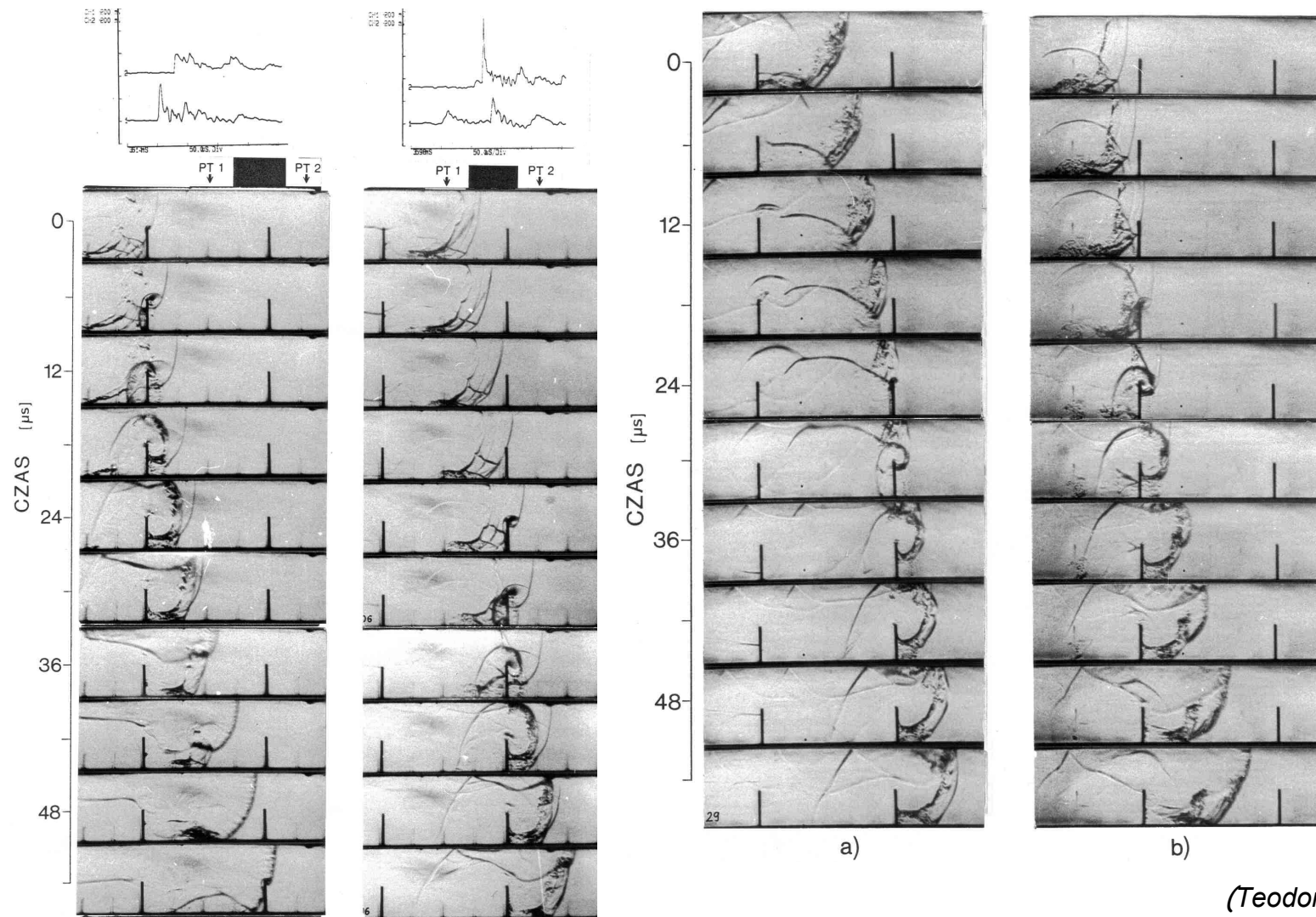
(Urtiev&Oppenheim,  
1965)

# DDT in smooth tube



(Urtiev&Oppenheim, 1965)

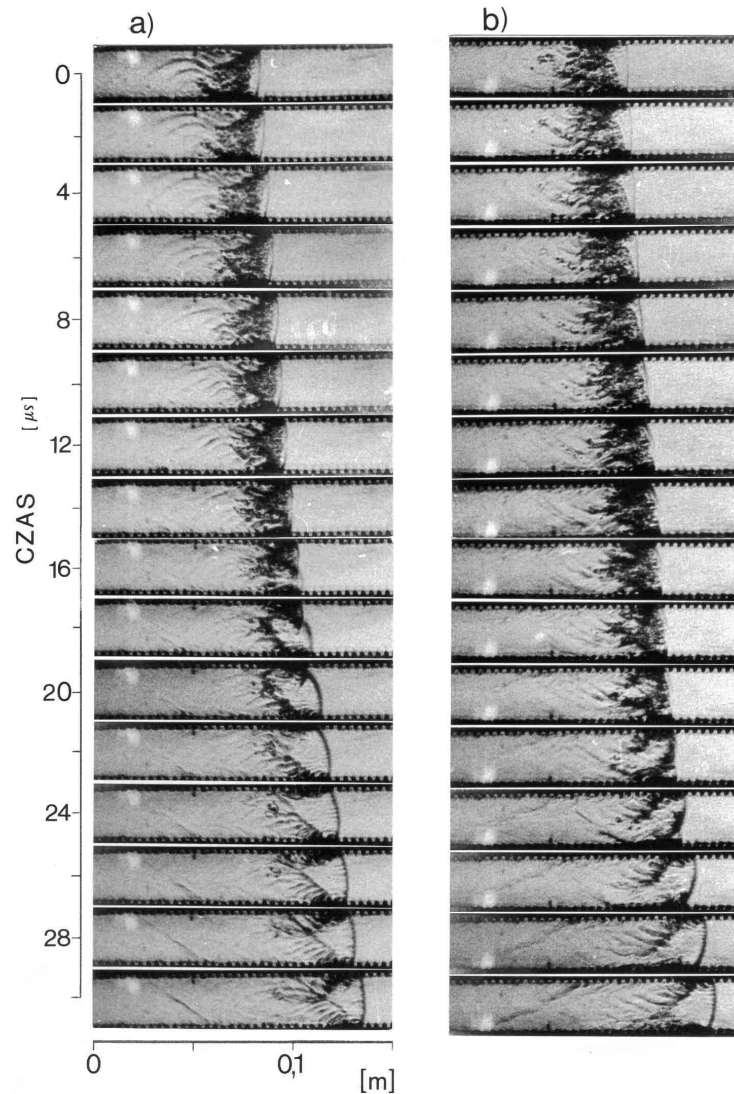
# DDT in tube with obstacles



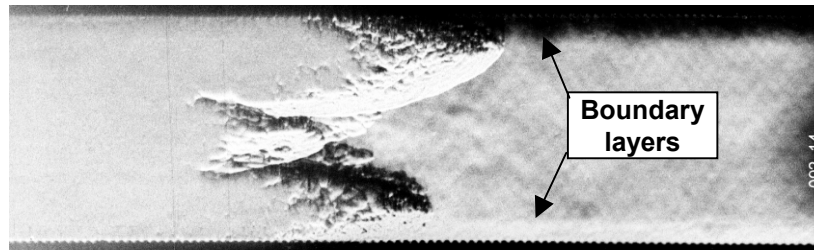
*(Teodorczyk, et al..1988)*



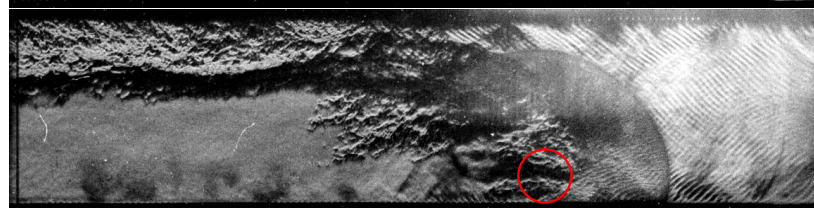
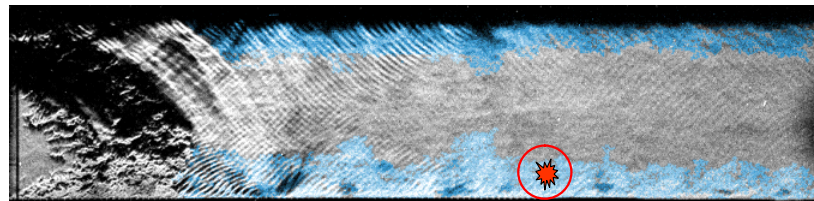
# DDT in rough channel



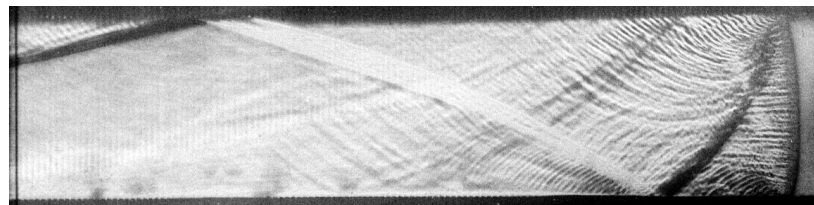
(Teodorczyk, 1990)



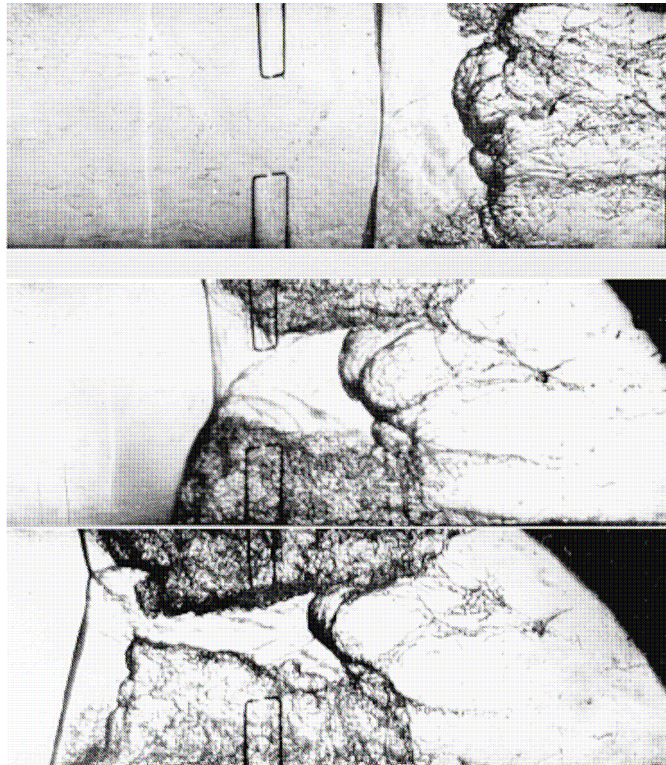
Flame speed 320 m/s



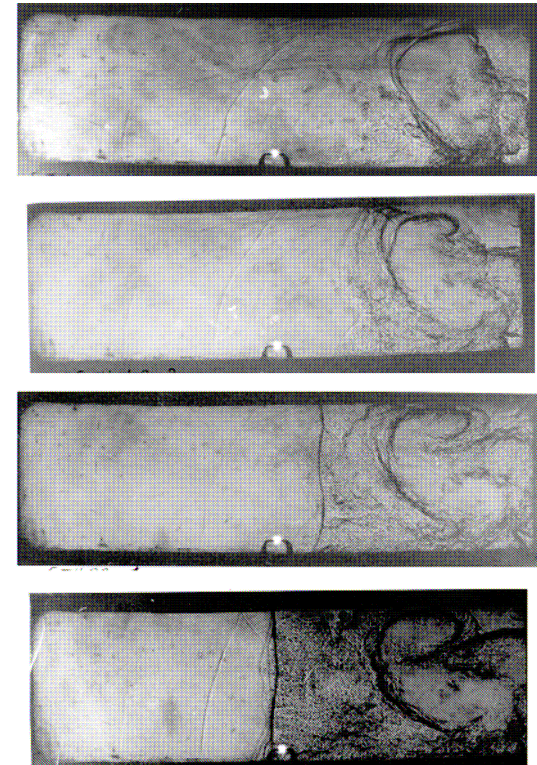
$p_0=0.55$  bar, 1090-1320 mm  
from ignition



(Kuznetsov M., Dorofeev S., 2005)



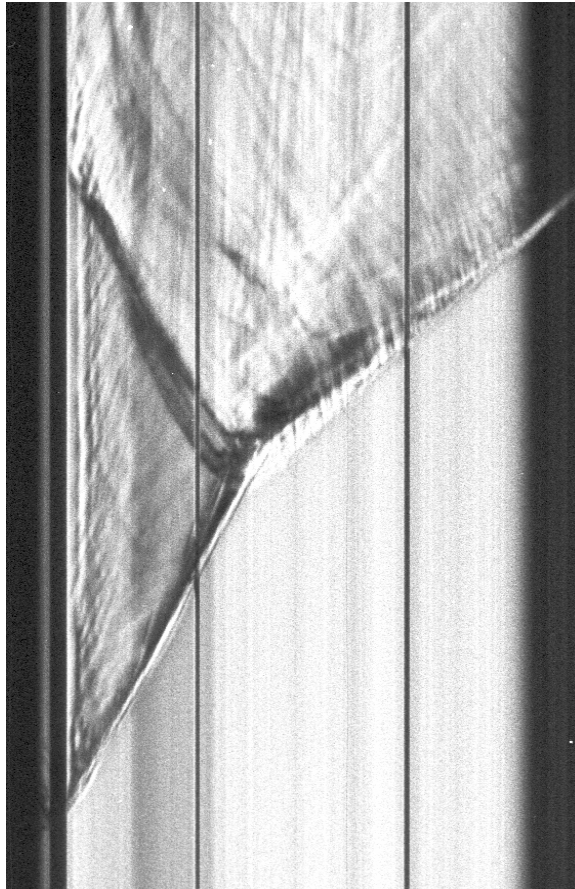
Reflected shock (*moving right to left*) emerging following multiple-shock flame interaction. Original incident shock Mach No. 1.7 (incident not shown). Mixture  $C_2H_4 + 3O_2 + 4N_2$ , initial pressure 13.2 kPa,  $\Delta t$  50  $\mu s$



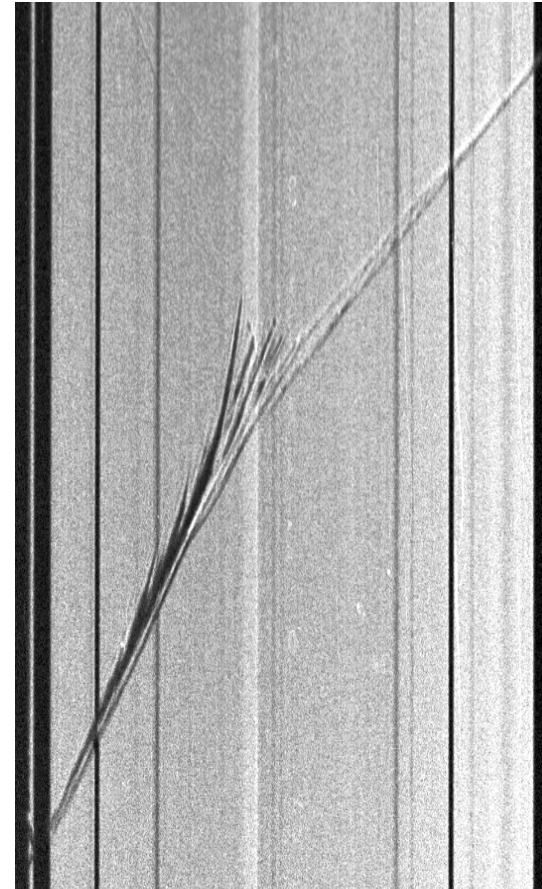
DDT resulting from the interaction of a reflected shock with a flame kernel

(Bombrey&Thomas, 2002)

# Two modes of DDT



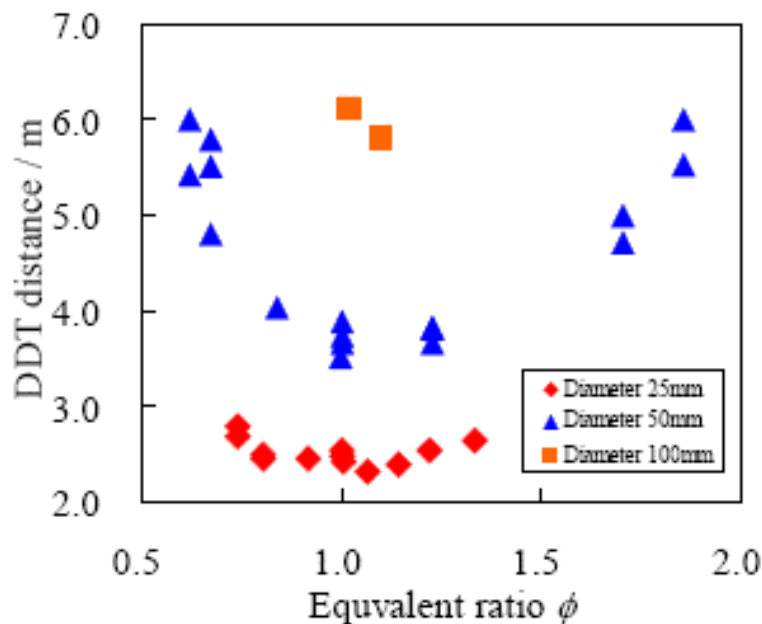
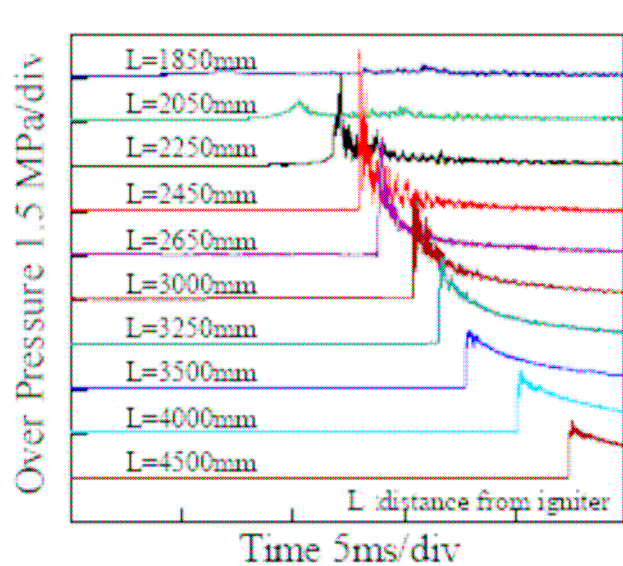
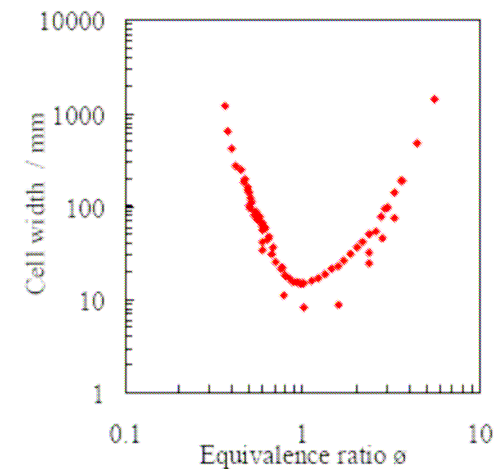
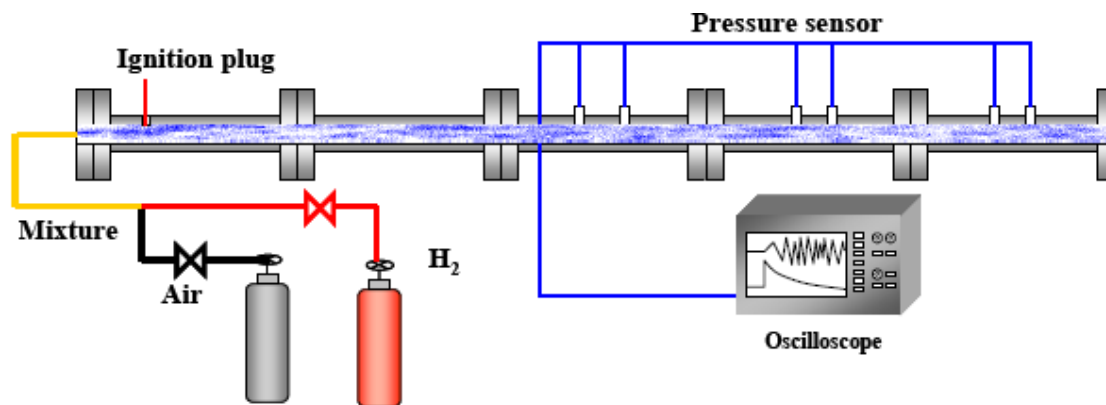
DDT via local explosion



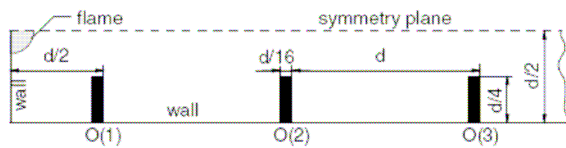
DDT via gradual amplification of  
transverse waves

*(Chao et al., 2002)*

# DDT of hydrogen-air in smooth tube

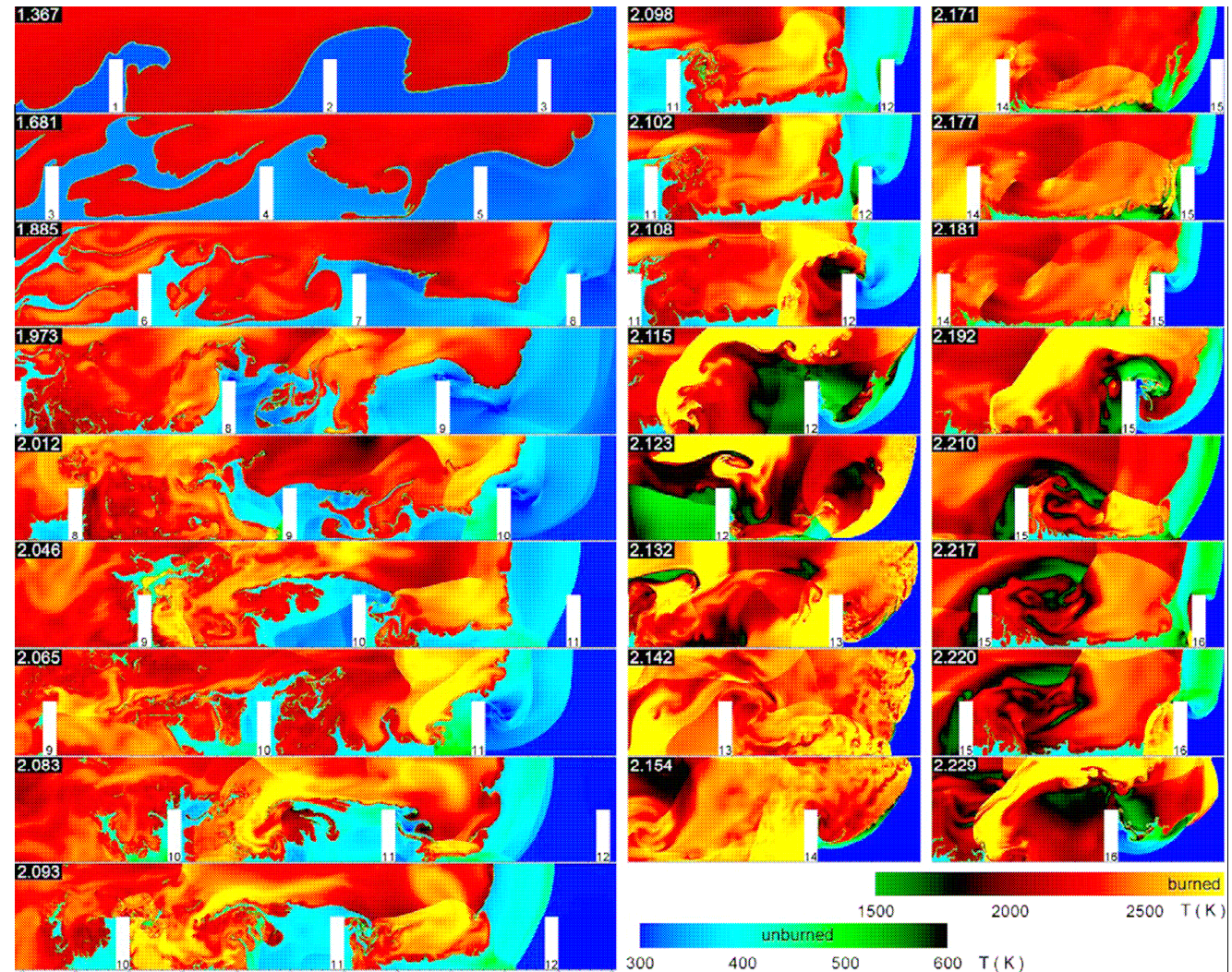


Source:  
Aizawa et al.,  
21st ICEDERS, July 23-27,  
2007, Poitiers



V.Gamezo et al., 31st  
Symposium International  
on Combustion,  
Heidelberg 2006

- stoichiometric hydrogen-air mixture at 0.1 MPa
- Channel with obstacles  
1m × 11cm × 2cm



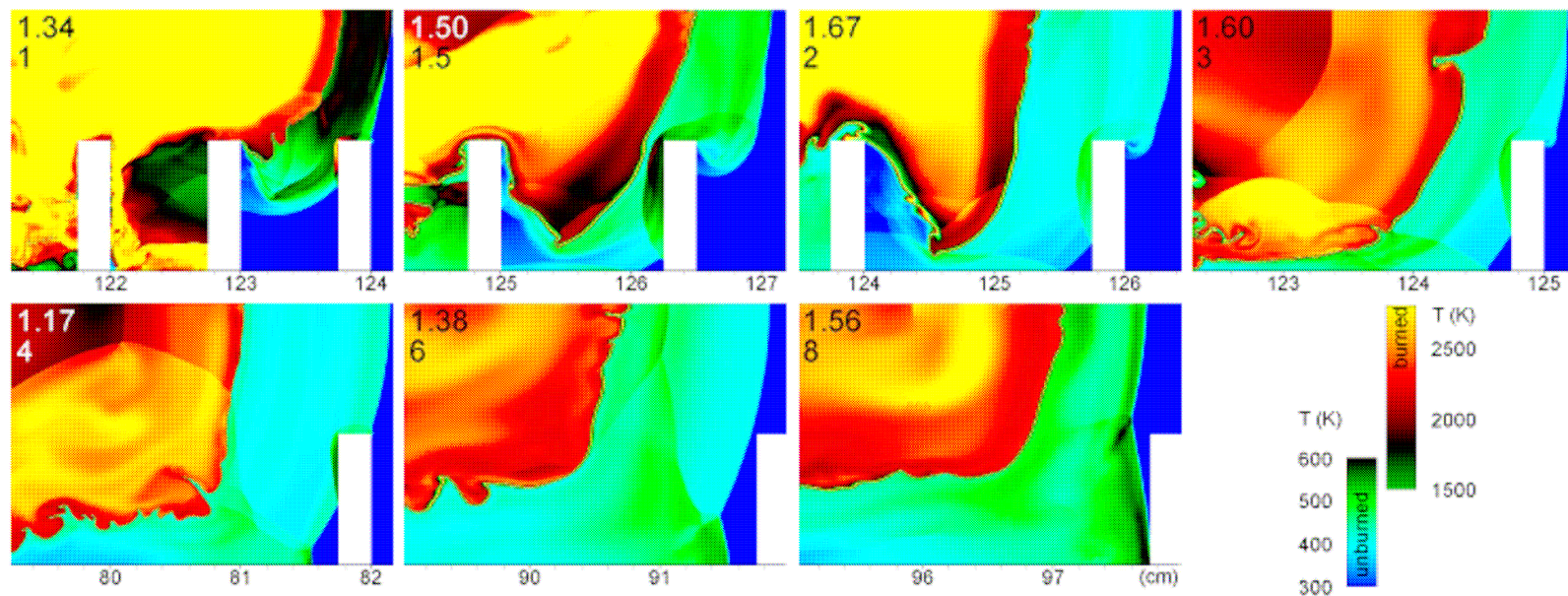


Fig. 2. Flame and shock configurations just before the detonation initiation ( $S=1, 4, 6, 8$ ) or at the end of the channel ( $S=1.5, 2, 3$ ). Time in milliseconds and  $S$  in centimeters are shown in frame corners.  $dx_{min}=1/512$  cm ( $S=1, 4$ ) or  $1/128$  cm ( $S=1.5, 2, 3, 6, 8$ ).

Source:  
 Gamezo et al..  
 21st ICDERS, July 23-27, 2007, Poitiers